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INHIBITED SHAPED CHARGE LAUNCHER TESTING OF SPACECRAFT SHIELD DESIGNS

Prepared by

Donald J. Grosch Southwest Research Institute

FINAL REPORT Contract No. NAS8-40634 SwRI Project No. 06-7698

Prepared for

NATIONAL AERONAUTICS and SPACE ADMINISTRATION
MARSHALL SPACE FLIGHT CENTER
Huntsville, Alabama

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APPROVED:

Charles E. Anderson, Jr., Ph.D., Director Engineering Dynamics Department



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EXECUTIVE SUMMARY

This report describes a test program in which several orbital debris shield designs were impact tested using the inhibited shaped charge launcher facility at Southwest Research Institute. This facility enables researchers to study the impact of one-gram aluminum projectiles on various shielding designs at velocities above 11 km/s. A total of twenty tests were conducted on targets provided by NASA-MSFC. This report discusses in detail the shield design, the projectile parameters and the test configuration used for each test. A brief discussion of the target damage is provided, as the detailed analysis of the target response will be done by NASA-MSFC.

1.0 INTRODUCTION

This report describes a test program in which several orbital debris shield designs were impact tested using the inhibited shaped charge launcher (ISCL) facility at Southwest Research Institute (SwRI). The ISCL facility enables researchers to study the impact of one-gram projectiles on various shield designs at velocities above 11 km/s. A total of twenty tests were conducted on targets provided by NASA-MSFC.

2.0 BACKGROUND

The basis for the ISCL is a metal-lined explosive cavity, referred to as a shaped charge. This device generates a long, plastically-deforming jet of material that travels at high speeds. Shaped charges, which have great penetrating capabilities, have been used for many years in anti-armor warheads and as oil well perforators.

The ISCL isolates the high-speed jet tip of the shaped charge through the use of an inhibitor. The inhibitor is placed within the cavity of the shaped charge. It allows the jet tip to develop as usual but prevents the remainder of the jet from forming. The isolated jet tip is the projectile used to simulate space debris.

The concept of an inhibited shaped charge launcher was first examined in the early 1960's. A re-examination of this concept began in 1987, when NASA-JSC funded SwRI to develop an explosive launcher for simulation of orbital impacts. Since then, several programs have been conducted by SwRI (one funded internally by SwRI, one by DNA, and the remainder by NASA) to refine the explosive launcher concept.

To utilize the explosive launcher in an environment that simulates the conditions in space, an evacuated hypervelocity launcher facility was designed and fabricated at SwRI. This facility increased the usefulness of the ISCL as a testing instrument by providing the means for conducting impact tests within a vacuum. It was designed to hold targets of various sizes and configurations so that different shield concepts could be tested.

3.0 TEST PROCEDURES

All tests were performed within the SwRI inhibited shaped charge launcher facility (Figure 1). Reduced pressures between 4 and 6 Torr were used in the target chamber.

An aluminum (1100-O) lined shaped charge with a 30° included angle was used for each test. Octol 70/30 was cast upon the aluminum liner to form the charge. The charge was initiated using an explosive bridge-wire detonator (EBW) and a precision initiation coupler (PIC). An OFHC copper inhibitor was used for each test to inhibit the formation of the shaped charge jet. (The reader is referred to Reference 1 for a detailed description of the ISCL concept.) Figure 2 shows a shaped charge with an inhibitor placed inside the aluminum liner.

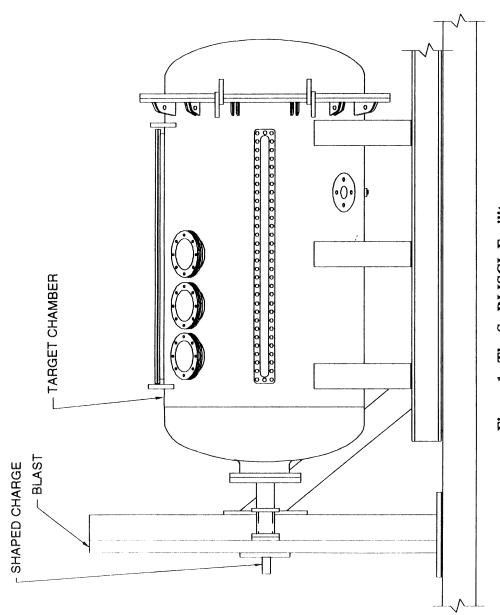


Figure 1. The SwRI ISCL Facility.

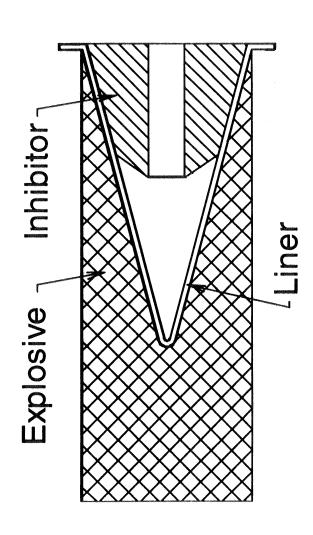


Figure 2. Shaped Charge with Inhibitor.

Flash x-ray (FXR) equipment was positioned to take radiographs of the projectile and debris cloud at various positions in the target chamber. Kodak direct exposure film (DEF) was used for all tests. The projectile geometry was measured from these radiographs and its velocity was calculated based on its position on the radiographs and the time at which the FXRs were taken.

The ISCL was configured to provide three (3) orthogonal views. The first FXR station used an HP 180 kV system with standard x-ray heads. The standard heads at this station produced low quality images of the projectiles, as the low-density aluminum projectiles do not absorb a large amount of this wavelength of x-ray. The image quality was enhanced during this program by placing an NDT-9 intensifier screen behind the film (intensifier screens are not required with the DEF film). This initial FXR station was used to determine the position-in-time and the integrity of the projectile.

The second and third FXR stations used HP 300 kV systems with soft x-ray heads. These soft heads create x-rays which are readily absorbed by the low-density aluminum material. Thus, a much clearer image of the projectile was produced with this type of system. The first soft FXR station was positioned to produce a radiograph of the projectile before it impacted the target. The projectile geometry and position-in-time were measured from this radiograph. The second soft FXR station, which was only used for the zero-degree obliquity tests, was positioned between the wall plate and the witness pack to provide a view of the post-impact debris cloud.

The FXR pulsers for each orthogonal view are triggered by time delay using Hewlett Packard (HP) Model 43114A digital delay generators. The delay generators are activated by the signal sent from the Reynolds FS-10 firing module to the detonator. Delay times were calculated prior to each test based on the position of the x-ray heads and the anticipated projectile velocity. Figure 3 shows the orientation for the pre-impact FXR stations.

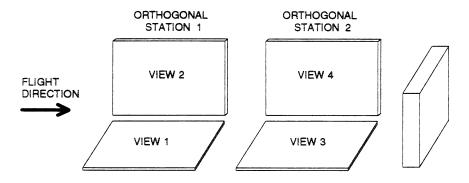


Figure 3. FXR Stations 1 and 2 Orientation.

During some tests, the third FXR station pulser was triggered using a make-screen. A make-screen consists of two pieces of aluminum-foil separated by a piece of mylar. A 700 volt potential is placed across the foils. However, since the Mylar is non-conductive, the current cannot flow from one foil to the other. Therefore, electronically the make-screen appears as an open switch. When a metallic object penetrates the screen, the switch is closed and the current flows from one foil to the other. This flow of current (or closing of the switch) can be detected

and recorded by electronic monitoring equipment and the signal can be used to trigger devices such as the FXR equipment. For certain tests, a make-screen was positioned such that the leading particle behind the target wall would penetrate it and trigger the behind-wall FXR station. Since it does not rely on a calculated delay time, such a setup insured that this FXR station would be triggered at the proper time.

Photographs (35mm) were taken before and after each test to show the target configuration using Kodak ASA-200 color film. Additional photographs were taken of the front and back faces of each individual target plate after they had been impacted. These photographs have been sent to the technical monitor at MSFC.

Data sheets were filled out during the conduct of each. Information recorded on these sheets includes test date and number, inhibitor geometry, x-ray delay times and distances, and vacuum pressure. Copies of these data sheets are provided in this report as Appendix A.

Post test information is also included on these data sheets. This type of information includes the projectile geometry, orientation, mass, and velocity. This information is acquired by examining the radiographs on a back-lit digitizing table. Values such as geometry, orientation, and velocity are provided directly using a software program called FILM developed at SwRI. This program allows immediate measurements of the projectile data using simple, yet extremely accurate, calibration techniques that are implemented with the ISCL. (The reader is referred to Ref. 2 for more information on this calibration system).

A summary of the projectile geometry measurements is provided in Table 3 (see Appendix B). In this table, a total angle (pitch and yaw) is provided with a quadrant value. This quadrant is the location in which the projectile is angled towards. The quadrant numbering system is that of the standard Cartesian coordinate system and is taken looking at the impact surface of the target from the charge (Quadrant 1 being in the upper right-hand corner with numbers increasing counter-clockwise). When a target was tested at other than 0° obliquity, the target was first positioned on a level surface and then rotated clockwise (looking down on the target) to achieve the proper angle. When this was done, quadrants 1 and 4 were closer to the charge than quadrants 2 and 3.

Since the ISCL projectile is not a sphere or a perfect rod, some estimations are made when determining its mass. To provide the best possible estimate of projectile mass, the projectile for each test was analyzed individually. The assumptions made and the analysis done for each test is provided as Appendix C of this report. In most cases, values for the projectile inner diameter, outer diameter, length, L/D ratio, and total inclination angle (yaw and pitch) are provided based on the projectile shape.

4.0 TEST RESULTS

Testing began on 22 February 1996. The test matrix originally prepared by MSFC was followed until we had a failure of the shaped charge device on Test 7698-14. During this test, the RP-87 detonator fired as usual, but failed to detonate the PIC. This resulted in a large portion of

the Octol charge being damaged. Although the aluminum liner was not damaged, the charge itself could no longer be used for an ISCL test.

The supplier of the PIC was contacted in an attempt to determine the cause of the failure. The company stood by their original response that the RP-87 detonator was sufficient to detonate their PIC. Since SwRI had successfully used this detonator / PIC combination 28 times before, we agreed with their response and continued testing.

During the next two tests (Tests 7698-15 and 7698-16), the detonation train of the ISCL worked successfully. However, during Test 7698-17, the PIC again did not detonate. The program was stopped until a solution could be determined. After several discussions with the PIC manufacturer, we decided to implement a more powerful detonator. It was decided that since the detonation train worked most of the time, but not all the time, that our detonator must be at the threshold of working with that PIC design.

Several very fortunate situations simultaneously occurred that allowed the two failed tests to be repeated. First, the liners were not damaged by the PIC failures. Second, SwRI had another set of liners being prepared to be explosively loaded at the time of the failures. Third, NASA-JSC had a few extra ISCL charges in storage at SwRI that they loaned to NASA-MSFC so the test series could be completed.

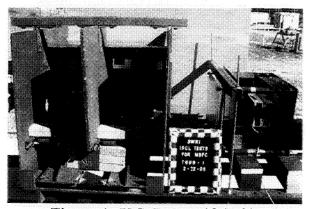


Figure 4. U.S. Lab at 0° (LO).

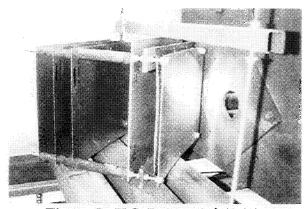


Figure 5. U.S. Lab at 45° (L45).

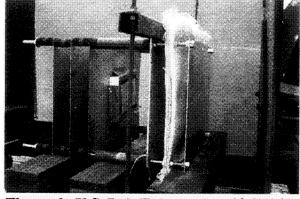


Figure 6. U.S. Lab Enhanced at 0° (LEO).

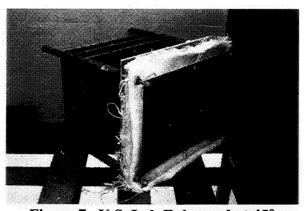


Figure 7. U.S. Lab Enhanced at 45°

Testing resumed on 12 March 1996 with Test 7698-18. The final five tests (Tests 7698-18 through 7698-22) were completed in two days. Table 1 provides detailed information

about the targets that were impacted during this test program. Figures 4 through 7 show several types of targets that were tested. The nomenclature for the targets is that the initial plate is the "face plate," followed by either a fabric layer (consisting of Nextel and Kevlar) or an MLI layer (which consists of several layers of multi-layer insulation), the second plate is the "wall plate," which is followed by the witness pack.

Table 1. Target Descriptions

TARGET NAME	ABREVIATION	TARGET DESCRIPTION
Witness Pack (For ALL Tests)	NONE	6" Space
		0.020" Al Plate
		2" Space
		0.020" Al Plate
		2" Space
		0.020" Al Plate
U.S. LAB	LO	0.032" Al 6061-T6
2/3 Scale	and	1.500" Space
0°, 45°, and 65° Obliquity	L45	20 Layers MLI
	and	1.500" Space
	L65	0.125" Al 2219-T87
		Witness Pack
U.S. LAB	L45F	0.050" Al 6061-T6
Full Scale		2.250" Space
45° Obliquity		20 Layers MLI
		2.250" Space
		0.188" Al 2219-T87
		Witness Pack
U.S. LAB ENHANCED	LE0	0.050" Al 6061-T6
2/3 Scale	and	1.500" Space
0° and 45° Obliquity	LE45	4 Layers NEXTEL
		4 Layers KEVLAR
		1.500" Space
		0.125" Al 2219-T87
		Witness Pack
U.S. LAB ENHANCED	LE0.8	0.063" Al 6061-T6
0.8 Scale	and	1.800" Space
0° and 45° Obliquity	LE45.8	5 Layers NEXTEL
		5 Layers KEVLAR
		1.800" Space
		0.150" Al 2219-T87
		Witness Pack
U.S. LAB ENDCONE	LEC0	0.032" Al 6061-T6
2/3 Scale	and	1.000" Space
0° and 45° Obliquity	LEC45	20 Layers MLI
		4.810" Space
		0.125" AI 2219-T87
		Witness Pack

TARGET NAME	ABREVIATION	TARGET DESCRIPTION
U.S. LAB ENDCONE	LEC0F	0.050" Al 6061-T6
Full Scale	and	1.500" Space
0° and 45° Obliquity	LEC45F	20 Layers MLI
		7.220" Space
		0.188" Al 2219-T87
		Witness Pack
JEM	JEM0	0.032" Al 6061-T6
2/3 Scale	and	1.500" Space
0° and 45° Obliquity	JEM45	20 Layers MLI
		1.500" Space
		0.080" Al 2219-T87
		Witness Pack
JEM	JEM45F	0.050" Al 6061-T6
Full Scale		2.250" Space
45° Obliquity		20 Layers MLI
		2.250" Space
		0.125" Al 2219-T87
		Witness Pack
U.S. LAB REAR WALL	LRW0	0.032" Al 6061-T6
2/3 Scale	and	1.500" Space
0° and 45° Obliquity	LRW45	20 Layers MLI
		1.500" Space
		0.125" Al 5456
		Witness Pack

NOTE: Space values given are approximate. The addition of the two space values is the true distance between the front of the face plate and the rear of the wall plate.

General target damage measurements were made by SwRI subsequent to the tests. These measurements include a rough sketch of the holes in both the face plates and the wall plates for each test. Rough measurements, taken with a tape measure, are provided on these sketches and in Table 2, which is a brief summary of each test. These sketches and measurements are provided as Appendix C.

MSFC also requested that specific measurements be recorded for the wall plate holes. To make these measurements, the wall plate of interest was positioned onto a piece of gridded paper. The hole in the plate was traced onto the paper keeping the pencil perpendicular to the paper and in contact with the side of the hole. Any cracks that might occur near the impact hole were also traced and their lengths recorded. Finally, the longest distance between any two cracks was recorded as the tip-to-tip crack length. The tip-to-tip crack length values for each applicable test are provided in Table 4 of Appendix D, as are copies of the hole sketches. The detailed analysis of the target plates will be done by MSFC. The target materials were shipped back to the MSFC technical monitor shortly after the conclusion of the test program. Following are brief descriptions of each test.

Table 2. Test Summary

TEST NO.	MASS (g)	VELOCITY (km/s)	TARGET	WALL DAMAGE (TIP-TO-TIP LENGTH)	COMMENTS
7698-1	1.35	11.68	L0	1" Dia. Hole (2.7")	Cookie-cut hole. Very symmetric debris pattern.
7698-2	1.12	11.38	LEC0	3.5" to 4.3" Dia. Hole (5.7")	Slightly "C" shaped projectile.
7698-3	0.99	11.64	LE0	3" to 4" Dia. Hole (11")	Typical projectile.
7698-4	0.94	11.37	L45	4.5" to 7" Dia. Hole (7.5")	Typical projectile.
7698-5	1.03	11.77	JEM0	4.5" to 5" Dia. Hole (8.3")	Typical projectile.
7698-6	0.84	11.37	LRW0	4.5" Dia. Hole (5.3")	Typical projectile.
7698-7	0.99	11.37	LEC45	4.7" to 6.3" Dia. Hole (3.6")	Typical projectile.
7698-8	0.82	11.40	LE45	No Penetration (-)	Poorly Formed Small Projectile
7698-9	0.97	11.42	JEM45	4" to 6.3" Dia. Hole (13")	Typical projectile.
7698-10	1.42	11.35	LRW45	1.5" Dia. Hole (3.5")	Typical projectile.
7698-11	1.28	11.47	LEC0F	0.5" to 1.25" Hole (3.1")	Typical (but longer than usual) projectile. Near the Ballistic Limit.
7698-12	0.76	11.45	LE0.8	No Penetration (-)	Poorly Formed Small Projectile
7698-13	1.04	11.51	L45F	0.625" Dia. Hole (0.8")	Typical projectile.
7698-14	-	-	LE0.8	- (-)	Detonator Went, PIC Did Not
7698-15	-	-	LE0.8	- (-)	Projectile Hit Stripper #2 Bad Shot
7698-16	1.09	11.51	LE45	No Penetration (-)	Odd Projectile, Had a Flat Impact due to Mass on Side

TEST NO.	MASS (g)	VELOCITY (km/s)	TARGET	WALL DAMAGE (TIP-TO-TIP LENGTH)	COMMENTS
7698-17	-	-	L65	- (-)	Detonator Went, PIC Did Not
7698-18	0.97	11.30	L65	No Penetration (-)	Typical projectile.
7698-19	1.11	11.32	L450.8	0.75" to 2" Dia. Hole (4.8")	Used an 0.050" 6061-T6 Face Sheet.
7698-20	1.23	11.38	L45	2.75" to 1.25" Dia. Hole (6")	Projectile is an Open Cylinder (cross section is "C" shaped)
7698-21	-	-	LE0.8	1.5" to 3.5" Dia. Hole (7")	Equipment Failure - No Pre-Impact Radiographs. Based on target, assume a fairly lethal projectile (high L/D, flying straight).
7698-22	0.98	11.36	JEM45F	0.5" to 1" Dia. Hole (6.8")	Projectile is not completely closed along axial length. About 1/3 of perimeter is opened (somewhat "C" shaped).

TEST 7698-1

This test resulted in a slightly larger than average (1.35 gram) projectile. An elliptical hole, roughly 15.2 by 10.2 cm (6 by 4 inches), was produced in the face sheet. This hole was jagged and the plate petalled in several places. The MLI layer was shredded, as was the case in each test in which the material was used. A 2.5 cm (1 inch) diameter hole was produced in the wall of the target. This was a "cookie-cut" hole, as there were no petals. A large hole resulted in the first two witness plates and the third witness plate was severely deformed with many small holes.

The third FXR station for this test was configured to be triggered using a make-screen. However, although the screen was working prior to test, during the evacuation of the target chamber it malfunctioned. Therefore, the station was triggered using a calculated time delay. A very nice post-impact debris pattern was captured during this test. Contact-prints of these radiographs are shown in Appendix E. Note that the make-screen is visible in each radiograph.

TEST 7698-2

The projectile for this test was cylindrically shaped with about one-fifth of the perimeter missing (shaped like the letter "C"). Its mass was right at the average value of 1.12 grams. The hole in the face plate had a diameter from about 8.9 to 10.7 cm (3.5 to 4.2 inches) with one large petal that extended to about 10.6 cm (4.2 inches) from the estimated impact point. The hole in the wall plate was typically 3.1 cm (1.2 inches) in diameter with one petal widening the hole to 5.6 cm (2.2 inches) wide. Witness Plate 1 had a roughly 15.2 cm (6 inch) hole with several smaller holes around it. Plate 2 had several holes and Plate 3 had only a few holes. The behindwall plate radiographs reveal a very concentrated debris pattern.

TEST 7698-3

The projectile in this test has a slightly longer L/D ratio than average (2.3), but is about the average mass (.99 grams). The hole in the face plate is about 7.6 cm (3 inches) in diameter and extends to 8.9 cm (3.5 inches) and 10.2 cm (4 inches) at two petal locations. The wall plate has a jagged hole that varies between 8.9 and 12.7 cm (3.5 and 5.0 inches) in diameter and up to 19.1 cm (7.5 inches) at a petal. All three witness plates had multiple small holes and rearward deformation.

TEST 7698-4

This was the first 45° obliquity test. The aim point for this and all oblique-impact tests was 5.1 cm (2 inches) forward horizontally from the center of the wall plate. This modification of the impact point insured that all plates would be impacted during the high-obliquity tests. As for all oblique impacts, the third (behind-wall) FXR station was not used for this test.

The projectile for Test 7698-4 had an L/D of 1.98 and a mass of 0.94 grams. The face plate had a roughly 11.4 cm (4.5 inch) hole that extended to as much as 17.8 cm (7 inches) at one

point. The wall plate had a roughly 6.4 cm (2.5 inch) diameter hole, and cracks ran left and right 10.2 and 6.4 cm (4 and 2.5 inches) respectively. Cracks that extended to the edge of the plate ran up and down such that the plate was cut in two. Damage on the witness plates decreased from many small holes on the front plate to only a few on the third plate.

TEST 7698-5

Test 7698-5 was a 0° obliquity shot. It produced a very jagged hole in the face plate with diameter values that varied from 11.4 to 17.0 cm (4.5 to 6.7 inches). The wall plate had a nominally 6.4 cm (2.5 inch) diameter hole with several petals that extended the hole size up to 12.7 cm (5 inches) long. The first witness plate has an 6.4 cm (8 inch) diameter hole and a large deflection. Plate #2 was fractured into two pieces with a large hole and deflection. The third witness plate is largely deformed and has several small holes in it. The behind-wall radiographs show a dispersed debris pattern.

TEST 7698-6

A relatively small (.84 gram) projectile was produced during this test. It resulted in a jagged face plate hole that was up to 11.4 cm (4.5 inches) in diameter. The wall had a 7.6 cm (3 inch) diameter hole with a single petal that extended the radius to 6.8 cm (2.7 inches). Witness Plate #1 had a large hole and deformation which decreased in Plate #2. Plate #3 had only a few small holes. The behind-wall radiographs show a debris cloud consisting of mainly small particles.

TEST 7698-7

This test was another 45° shot. A low L/D (1.15) projectile with a mass of 0.99 grams was produced. The face plate hole was about 11.9 cm (4.7 inches) in diameter and the longest petal extended the hole radius to 15.7 cm (6.2 inches). The hole in the wall plate was around 3.8 cm (1.5 inches) in diameter with no petalling. Damage was minimal to Witness Plate #1 and Plate #2 was not damaged.

TEST 7698-8

This test produced a projectile that was not well formed. It appeared to be a cylinder that consisted of only about half of the cylinder wall. Although a poor shape, it had a substantial mass (0.82 grams). The projectile produced an elliptical hole in the face plate that measured 6.4 by 10.2 cm (2.5 by 4.0 inches). The wall plate was not penetrated but was bulged a few centimeters.

TEST 7698-9

The projectile for Test 7698-9 produced a very jagged hole which resulted in large petal formation in the face plate. The diameter varied from 10.2 to 15.7 cm (4.0 to 6.2 inches). A large amount of damage was also done to the wall plate, which was broken in two pieces by the

impact. The approximate hole size of the wall plate was 15.2 cm (6 inches). A central hole was produced in all three witness plates. The damage decreases from many small holes in Witness Plate #1 to only a few small holes in Plate #3.

TEST 7698-10

A larger than average projectile (1.42 grams) was produced during this test. The face sheet had an almost rectangularly-shaped hole with a minor diameter of 6.4 cm (2.5 inches) and a major diameter of 14.0 cm (5.5 inches). The wall plate had a 3.8 cm (1.5 inch) diameter hole with no petals. Witness Plate #1 had a large 12.7 cm (5 inch) diameter hole in it. This damage decreased to multiple small holes by Plate #2 and only a few small holes by Plate #3.

TEST 7698-11

A high L/D ratio (3.4) projectile was produced during this test. It produced a hole with a 8.9 to 10.7 cm (3.5 to 4.2 inch) diameter in the face sheet. The damage to the wall plate suggests the test was very near the ballistic limit of the material. It appears that the small hole in the plate was caused by spalling of the rear surface of the plate, not by penetration. The hole, which is really a small crack, is about 1.3 cm (0.5 inches) wide by about 3.0 cm (1.2 inches) long. The resulting debris on the back side of the wall plate produced only small holes in the initial witness plate and only two small impacts on the second plate.

TEST 7698-12

A poorly-formed, low-mass (0.76 grams) projectile was produced during this test and resulted in no penetration of the target wall plate. The projectile formed a roughly 5.1 cm (2 inch) diameter hole in the face plate with no petalling. About a 2.5 cm (1 inch) bulge resulted at the impact point on the wall.

TEST 7698-13

A very nice projectile was produced during Test 7698-13. It produced a hole in the face plate that varied between 8.1 and 11.4 cm (3.2 and 4.5 inches) in diameter and had three large petals. A "cookie-cut" hole approximately 1.5 cm (0.6 inches) in diameter resulted in the wall plate. Witness plate damage decreased from about nine small holes in the initial plate to two small holes in the third.

TEST 7698-14

During this test, the detonator fired while the PIC did not. The result was that the charge could not be fired.

TEST 7698-15

The projectile formed during this test did not fly straight and it impacted the second stripper plate. The impact occurred approximately 1.3 cm (0.5 inches) from the edge of the

stripper hole. The result of this near miss was that a large amount of steel spall (from the stripper plate) and projectile material impacted and destroyed the target.

TEST 7698-16

The projectile produced during this test was not typical. It appears that the projectile opened up severely and resembled a flat plat more than a hollow cylinder. The mass of 1.09 grams was still present, it was just in a non-typical form. The projectile created a 8.1 to 10.2 cm (3.2 to 4.0 inch) diameter hole in the face plate and only a 2.0 cm (0.8 inch) high bulge in the wall plate.

TEST 7698-17

Again, the detonator fired and the PIC did not. No shot occurred.

TEST 7698-18

This was the only 65° obliquity test performed during this program. A typically shaped projectile was produced but did not penetrate the wall plate. It did produce a jagged hole in the face plate that measured between 15.2 and 17.8 cm (6 and 7 inches) in diameter with one crack that ran to the bottom edge of the plate.

TEST 7698-19

Another very nice projectile was produced during this test. It produced a highly-petalled hole in the face plate. Hole diameters varied from 7.6 to 11.4 cm (3.0 to 4.5 inches). A "cookiecut" hole resulted in the wall plate. Its measured about 2.0 by 5.1 cm (0.8 by 2.0 inches). The test resulted in a large number of small holes in Witness Plate #1. The number of small holes decreased greatly by Plate #2 and only one hole was created in Witness Plate #3.

TEST 7698-20

The projectile for this test appeared to be an opened-up cylinder whose cross-section looked like the letter "C". It produced a jagged hole in the face plate that measured about 11.4 cm (4.5 inches) in diameter. The hole diameter increased to 15.7 cm (6.2 inches) at two locations where petals occurred. The wall plate had a hole with a nominal diameter of 4.3 cm (1.7 inches) that petalled in two places. The petalling of the wall plate increased the hole diameter to 6.9 cm (2.7 inches) at one point. All three witness plates had a main hole that was approximately 1.3 cm (0.5 inches) in diameter. Plate #1 had approximately 40 smaller holes that decreased to 5 by Plate #3.

TEST 7698-21

Due to a malfunction of the X-ray equipment, we did not get an image of the projectile during this test. Only the third FXR station triggered properly and produced a nice image of the behind wall debris pattern. The hole in the face plate was nominally 7.6 cm (3 inches) in

diameter and had only a small amount of petalling. The hole in the wall plate was 3.8 cm (1.5 inches) in diameter and had a single petal that extended the hole size to 8.9 cm (3.5 inches). Four long cracks ran from the hole in the wall plate. Witness plate damage was typical, with a large number of small holes in Plate #1 that reduced to a few small holes by Plate #3.

TEST 7698-22

The 0.98 gram projectile produced during the final test produced a 10.2 cm (4 inch) diameter hole in the face plate. One large and several small petals occurred in this plate. The wall plate had a "cookie-cut" hole with four large cracks running from it. The hole was about 1.3 by 2.5 cm (0.5 by 1.0 inches). The first and second witness plates had about eight small holes and one 1.0 cm (0.4 inch) diameter hole through them. The third witness plate only had a single 0.5 cm (0.2 inch) diameter hole.

5.0 SUMMARY

A total of twenty ISCL tests were performed on shield designs provided by NASA-MSFC. Flash x-rays were used to image the ISCL projectile before impact for projectile velocity and geometry measurements. In some tests, flash x-rays were used to image the behind-wall debris pattern. The average projectile mass was 1.05 grams and the average velocity was 11.45 km/s. Basic measurements of hole size and shape were made and are included in this report. Detailed analysis of the targets will be done by NASA-MSFC.

6.0 ACKNOWLEDGMENTS

SwRI would like to thank Ms. Jeanne Crews of NASA-JSC for her help in this program. She loaned us the initial twenty ISCL charges so that the program would not be delayed (there is a long lead time for procuring the ISCL charges). She also loaned us an additional two charges after the PIC failure tests. This generosity allowed us to complete the program in approximately 1/2 the estimated time.

7.0 REFERENCES

- 1. "Development of an Inhibited Explosive Hypervelocity Launcher," by D. Grosch, J. Walker, S. Mullin, and R. Tullos, Final Report, SwRI Project No. 06-3513, July 1991.
- 2. "Improved Photogrammetry at SwRI," by D. J. Grosch and J. P. Riegel, III, presented at the 44th Aeroballistics Range Association Meeting, Munich, Germany, September 1993.

APPENDIX A

Test Data Sheets

INHIBITED SHA	APED CHARGE LAUNCHER DATA SHE	ET
TEST NUMBER	7698-1 DATE 2-22-	96 (10:50 AM)
ISCL TEST NO	130 Sunny \$	85°F
EXPLOSIVE		
CHAR	rge number6	_
CHAF	RGE WEIGHT <u>657.6</u>	_ (grams)
DETC	NATOR TYPE RP-80	
INHIBITOR	1 2 5 6	
OVER	RALL HEIGHT /. 759	_ (in)
INNE	r diameter <u>0.469</u>	_ (in)
FXRs		
HEAD #1:	SIZE 150	
	DISTANCE FBOC 96	• ` ′
	DELAY TIME /27	\
HEAD #2:	SIZE 300 soft	1 2 04120
	DISTANCE FBOC 57	\int (in)
	DELAY TIME	_ ` ′
HEAD #3:	SIZE 300 soft	
	DISTANCE FBOC	(in)
	DELAY TIME 2/7 psec	_ (usec)
	JRE (torr)	,
TARGET DESCRI	PTION MSFC LO (US LAB, 2)	3 Scare, @ 0°)
COMMENTS _ F	ace Plane: Large Pettaces He	oce (26"p)
_ W	ALL RATE: Cookie-Cut Hole	(≈1"ø)
- W	Triess: Severe DAMAGE.	

	7698-2 DATE_	7-22.96
ISCL TEST NO	/3/	
EXPLOSIVE	11.1	
CHA	RGE NUMBER	and the second s
CHA		(grams)
DETO	ONATOR TYPE RP-87	
INHIBITOR	1769	
	RALL HEIGHT 1759	
INNE	R DIAMETER 0,468	(in)
FXRs		
HEAD #1:	SIZE	
	DISTANCE FBOCSAMe	
	DELAY TIMESAME	
HEAD #2:	SIZE 300 soft	
	DISTANCE FBOC SAME	
	DELAY TIME	
HEAD #3:	SIZE 300 soft	(kV)
	DISTANCE FBOC SAMP DELAY TIME Tryse of Make	(in) /" In fact of
205.2750-	DELAY TIME Trene of Make -	(usec)
TA CHILL DESCRI	$JRE \underline{\frac{4/2}{2}} (torr)$	

INHIBITED SHA	APED CHARGE LAUNCHER DATA	A SHEET
TEST NUMBER	7698-3 DATE_2	-23-96
ISCL TEST NO	7698-3 DATE 2 132 Hot	\$ Somy
		·
EXPLOSIVE	119	
CHAR	RGE NUMBER	-
CHAR	RGE WEIGHT 656.5	(grams)
DETO	NATOR TYPE <u>RP-87</u>	-
INHIBITOR	1759	
OVER	ALL HEIGHT	_ (in)
INNE	RALL HEIGHT 1,759 R DIAMETER 2,469	_ (in)
FXRs		
HEAD #1:	SIZE	_ (kV)
	DISTANCE FBOC	_ (in)
	DELAY TIME	_(usec)
HEAD #2:	SIZE 300 soft	
	DISTANCE FBOC	_ (in)
	DELAY TIME	_(usec)
HEAD #3:	DISTANCE FBOC	-(kV)
	DISTANCE FBOC	_(in) { U(b)
	DELAY TIME $\frac{\mathcal{O}_{1} + \mathcal{M}_{abc}}{\mathcal{A}_{1}} = 210$ RE	_(usec))
TARGET DESCRIE	PTION MSFC LEO	
		1 1 - 1
COMMENTS B.	g hold in Face Alato, Buy h	all on Parties
La	ye Petaled holoin Wall.	appeare that
- Land	une contributed to sella	
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^	ales & Pent sourced.	

	APED CHARGE		
TEST NUMBER	7698-4	DATE _	2-26-96
ISCL TEST NO	133		3000py 2 650
			/
EXPLOSIVE			
CHA	RGE NUMBER	# 41	
CHA	RGE WEIGHT	6 5 7 . 27	(grams)
DETO	ONATOR TYPE	RP-87	
INHIBITOR			
OVE	RALL HEIGHT	1.759	(in)
INNE	ER DIAMETER	8,469	(in)
FXRs			
HEAD #1:	SIZE	150	(kV)
	DISTANCE FBOC		
	DELAY TIME		(usec)
HEAD #2:	SIZE	300 soft	(kV)
	DISTANCE FBOC		(in)
	DELAY TIME		(usec)
HEAD #3:	SIZE	300 soft	(kV)
	DISTANCE FBOC		(in) Don't Use he was Andre
	DELAY TIME		(usec) Thouse thouse
VACUUM PRESSI	URE5	(tor	r) Laser
			k/ ."
Am Point	PTION MSFC	CENTER OF TA	1,521
Color Or	TRIGOT IS 7%.	Francisco State	
COMMENTS		-	

INHIBITED SH	APED CHARGE LAUNCHER DAT	A SHEET
TEST NUMBER	7698-5- DATE 2	-26-96
ISCL TEST NO	134	
EXPLOSIVE		
CHA	RGE NUMBER	_
CHA	RGE WEIGHT65 7. 0	_(grams)
DETO	ONATOR TYPE <u>RP-87</u>	_
INHIBITOR		
OVE	RALL HEIGHT	_ (in)
INNE	R DIAMETER	_ (in)
FXRs		
HEAD #1:	SIZE150	
	DISTANCE FBOC	(in)
	DELAY TIME	_ (usec)
HEAD #2:	SIZE 300 soft	
	DISTANCE FBOC	
	DELAY TIME	_ (usec)
HEAD #3:	SIZE300 soft	
	DISTANCE FBOC	(in)
	DELAY TIME THE SING 2/0	
VACUUM PRESSU	JRE 4//2 (torr)	
TARGET DESCRI	PTION JEMO	
		MAKE Screen
COMMENTS USE	2 WOODEN FIRME (3/4" PLYWOOD) T	8 SUPPORT
Cula re used Screen DID	1 styrofaam it always had	problems!

	APED CHARGE LAUNCHER DA	
EST NUMBER _	7698-6 DATE_	2-26-16
SCL TEST NO	135	
XPLOSIVE	# - C	
	RGE NUMBER	
CHA	RGE WEIGHT 652.3	(grams)
DET	ONATOR TYPE <u>RP-87</u>	MARCHANIA CORP.
NHIBITOR		
OVE	RALL HEIGHT	(in)
INNI	ER DIAMETER	(in)
XRs		
HEAD #1:	SIZE150	(kV)
	DISTANCE FBOC	(in)
	DELAY TIME	
HEAD #2:	SIZE 300 soft	(kV)
	DISTANCE FBOC	(in)
	DELAY TIME'(
HEAD #3.	SIZE 300 soft	
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	DELAY TIME Make Scie 201	1 + 4 puec
	DELAT TIME TWO STATES	(usec)
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ARGET DESCRI	IPTION MSFC LEWD	

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	APED CHARGE LAUNCHER DA	
TEST NUMBER	7698-7 DATE_	2-27-96
ISCL TEST NO	136	
EXPLOSIVE	10	
CHAF	RGE NUMBER	
CHAF	RGE WEIGHT 659,5	(grams)
DETC	NATOR TYPE <u>RP-87</u>	-
INHIBITOR	_	
OVER	RALL HEIGHT	(in)
INNE	r diameter0.468	(in)
FXRs		
HEAD #1:	SIZE150	(kV)
	DISTANCE FBOC	
	DELAY TIME	(usec)
HEAD #2:	SIZE 300 soft	
	DISTANCE FBOC	(in)
	DELAY TIME	(usec)
HEAD #3:	SIZE300 soft	
	DISTANCE FBOC	
	DELAY TIME	(usec)
VACUUM PRESSU	$\sqrt{RE} = \frac{4/2}{2} $ (torr)	
	PTION MSFC +7 LEC45	

TEST NUMBER 7698-8 DATE 2-27-76 ISCL TEST NO. 37 EXPLOSIVE CHARGE NUMBER /5 CHARGE WEIGHT 659.2 (grams) DETONATOR TYPE RP-87 INHIBITOR OVERALL HEIGHT 759 (in) INNER DIAMETER 0.469 (in) FXRS HEAD #1: SIZE 150 (kV) DISTANCE FBOC (in) DELAY TIME (usec) HEAD #2: SIZE 300 soft (kV) DISTANCE FBOC (in) DELAY TIME (usec) HEAD #3: SIZE 300 soft (kV) DISTANCE FBOC (in) DELAY TIME (usec) HEAD #3: SIZE 300 soft (kV) DISTANCE FBOC (in) DELAY TIME (usec) HEAD #3: SIZE 300 soft (kV) DISTANCE FBOC (in) DELAY TIME (usec) VACUUM PRESSURE (torr)		APED CHARGE LAUNCHER DA'	
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CHARGE NUMBER	ISCL TEST NO	37	
CHARGE NUMBER			
DETONATOR TYPE		1	
NHIBITOR OVERALL HEIGHT	CHA	RGE NUMBER	
OVERALL HEIGHT	CHA	RGE WEIGHT 657.2	(grams)
OVERALL HEIGHT	DET	ONATOR TYPERP-87	
HEAD #1: SIZE	NHIBITOR	_ 0	•
HEAD #1: SIZE	OVE	RALL HEIGHT	(in)
HEAD #1: SIZE	INNE	ER DIAMETERO, 469	(in)
DISTANCE FBOC			
DELAY TIME	HEAD #1:	SIZE150	(kV)
HEAD #2: SIZE		DISTANCE FBOC	(in)
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DISTANCE FBOC (in) DELAY TIME (usec) VACUUM PRESSURE (torr)		DELAY TIME	(usec)
DELAY TIME(usec) VACUUM PRESSURE(torr)	HEAD #3:	SIZE 300 soft	(kV)
VACUUM PRESSURE(torr)		DISTANCE FBOC	(in)
VACUUM PRESSURE(torr)		DELAYTIME	(usec)
	VACUUM PRESSI	URE $\frac{4^{1/2}}{2}$ (torr)	
TARGET DESCRIPTION MSFC LE 45	TARGET DESCRI	PTION MSEC LE45	
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TEST NUMBER	7698-1	DATE <i>2</i> _	- 28 - 96
ISCL TEST NO	/38	- Cank	50°F
EXPLOSIVE		11.0	
CHA	RGE NUMBER	# 7	_
CHA	RGE WEIGHT	655, 9	_ (grams)
DETO	ONATOR TYPE	RP-87	_
INHIBITOR		1 > -9	
OVE	RALL HEIGHT	1.759	_ (in)
INNE	ER DIAMETER	0,468	_ (in)
FXRs			
HEAD #1:		150	
	DISTANCE FBOC		(in)
	DELAY TIME	(/	_ (usec)
HEAD #2:	SIZE	300 soft	(kV)
	DISTANCE FBOC	L /	(in)
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HEAD #3:	SIZE	300 soft	(kV)
	DISTANCE FBOO	:	(in)
	DELAY TIME		(usec)
VACUUM PRESSI	URE 4/2	(torr)	

TARGET DESCRIPTION MSFC JEM 45

		LAUNCHER DATA	
TEST NUMBER	7698-10	DATE	-28-96
ISCL TEST NO	13.5	_ Cloudy	50°
EXPLOSIVE		_	
CHAR	RGE NUMBER	32	-
CHAR	RGE WEIGHT	5=7,8	(grams)
DETO	NATOR TYPE	RP-87	-
INHIBITOR		. 5.	
OVER	RALL HEIGHT	1,75 9 0.457	(in)
INNE	R DIAMETER	0.457	(in)
FXRs			
HEAD #1:	SIZE	150	_ (kV)
	DISTANCE FBOO	- (1	_ (in)
	DELAY TIME	/)	_(usec)
HEAD #2:		300 soft	
	DISTANCE FBOO		
	DELAY TIME		
HEAD #3:	SIZE	300 soft /	_(kV)
	DISTANCE FBOO		_ (in)
	DELAY TIME		_ (usec)
VACUUM PRESSU	RE 4/2	<i>5</i> (torr)	
TARGET DESCRI	PTION MSFC	LRW 45	_

INHIBITED SHAPED CHARGE LAUNCHER DATA SHEET				
	7698- DATE_3	3-2-96		
ISCL TEST NO	140 SD-7=	, Cloudy, 4:25 am		
		•		
EXPLOSIVE				
CHAI	RGE NUMBER			
CHAI	RGE WEIGHT 653, 3	(grams)		
DETO	NATOR TYPE <u>RP-87</u>			
INHIBITOR	1 - 2			
	RALL HEIGHT 1759			
INNE	R DIAMETER	(in)		
FXRs				
HEAD #1:	SIZE150	(kV)		
	DISTANCE FBOC			
	DELAY TIME	(usec)		
HEAD #2:	SIZE300 soft	(kV)		
	DISTANCE FBOC			
	DELAY TIME	(usec)		
HEAD #3:	SIZE300 soft	(kV)		
	DISTANCE FBOC	(in)		
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VACUUM PRESSURE 5/2 (torr)				
TARGET DESCRIPTION LECOF				
COMMENTS Very More The BALLISTIC LIMIT. SMALL Hole (5).				
COMMENTS V	ery were The BALLISTIC LIM	IT. SAME MORE/S).		
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	APED CHARGE LAU		-		
	7698-12		2-4-76		-
ISCL TEST NO		7	· · · · · · · · · · · · · · · · · · ·		
EXPLOSIVE		٦			
CHAI	RGE NUMBER	·-	_		
CHAI	RGE WEIGHT	1.8	_(grams)		
DETO	ONATOR TYPERP-87		_		
INHIBITOR	RALL HEIGHT	759	(in)		
DINE	R DIAMETER	(= =	_ (in)		
FXRs	R DIAMETER		_ (=-)		
HEAD #1:	SIZE150		_ (kV)		
	DISTANCE FBOC				
	DELAY TIME	//	_ (usec)		
HEAD #2:	SIZE300 s	oft	_ (kV)		
	DISTANCE FBOC	11	(in)		
	DELAY TIME	11	_ (usec)		
HEAD #3:	SIZE300	soft	(kV)	SO	
	DISTANCE ERCC	1)	(in)	MAGE	ے ہ
	DELAY TIME	SCRN SCRN	_(usec)	MAGE NO PEN WALL	
VACUUM PRESSU	JRE 4//2	(torr)			
TARGET DESCRI	PTION MSFC L	E0, 8			

COMMENTS NO WALL POLETESTION

INHIBITED SHA	APED CHARGE LAUNCHER DAT	A SHEET	
TEST NUMBER	7698-3 DATE 3	_ 4 _ 3 }	
ISCL TEST NO	/ 42		
EXPLOSIVE			
CHAR	RGE NUMBER	_	
CHAR	RGE WEIGHT	_ (grams)	
DETO	NATOR TYPE <u>RP-87</u>	_	
INHIBITOR			
OVER	ALL HEIGHT	_ (in)	
INNE	R DIAMETER	_ (in)	
FXRs			
HEAD #1:	SIZE150		
	DISTANCE FBOC		
	DELAY TIME		
HEAD #2:	SIZE 300 soft		
	DISTANCE FBOC		
	DELAY TIME		
HEAD #3:	SIZE 300 soft		
	DISTANCE FBOC		
	DELAY TIME	_ (usec)	
VACUUM PRESSURE (torr)			
TARGET DESCRI	PTION MSFC L45F		

	IAPED CHARGE		
TEST NUMBER _	7698-10	5 /4 DATE	3-5-96
ISCL TEST NO	14 143		
EXPLOSIVE		16	
CHA	RGE NUMBER	10	
СНА	RGE WEIGHT	655.6	(grams)
DET	ONATOR TYPE	RP-87	
INHIBITOR			
OVE	RALL HEIGHT	1, 150	(in)
INNI	ER DIAMETER	0,468	(in)
FXRs			
HEAD #1:	SIZE	150	(kV)
		h	
	DELAY TIME	1,	(usec)
HEAD #2:	SIZE	300 soft	(kV)
	DISTANCE FBOC	<i>f</i> ¹	(in)
	DELAY TIME	1,	(usec)
HEAD #3:	SIZE	300 soft	(kV)
	DISTANCE FBOC	11	(in)
		AKE SCRN	(usec)
VACUUM PRESSI	URE 4/2	(torr)	
TARGET DESCRI	PTION MSFC	LE0.8	
TARGET DESCRI	Prion MsFc	LE0.8	

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INHIBITED SH	APED CHARGE LAUNCHER DAT	A SHEET	
	7698- 5 DATE 3	5-76	
ISCL TEST NO	144		
EXPLOSIVE			
CHAI	RGE NUMBER = 3	_	
CHAI	RGE WEIGHT	_(grams)	
DETO	ONATOR TYPERP-87	_	
INHIBITOR			
OVE	RALL HEIGHT 1.750	_ (in)	
INNE	R DIAMETER	_ (in)	
FXRs			
HEAD #1:	SIZE150	_ (kV)	
	DISTANCE FBOC//	(in)	
	DELAY TIME	_ (usec)	
HEAD #2:	SIZE		
	DISTANCE FBOC	(in)	
		_ (usec)	
HEAD #3:	SIZE300 soft	_ (kV)	
	DISTANCE FBOC		
	DELAY TIME MANE SCR		
VACUUM PRESSURE 4//2 (torr)			
TARGET DESCRI	PTION MSFC LED.8 (Plen	ent #14)	
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COMMENTS	rosection ++ STRIPE #	•	
7.	rojection # STRIFT #2 roject (FAIL F FAE NO)	of a second	

INHIBITED SHAPED CHARGE LAUNCHER DA			
TEST NUMBER 7698 - 16 DATE	3-5-79		
ISCL TEST NO.			
EXPLOSIVE			
CHARGE NUMBER	- Constantin		
CHARGE WEIGHT	(grams)		
DETONATOR TYPERP-87	NAME OF THE PROPERTY OF THE PR		
INHIBITOR			
OVERALL HEIGHT	(in)		
OVERALL HEIGHT 1.7 60 INNER DIAMETER 0.493	(in)		
FXRs			
HEAD #1: SIZE150	(kV)		
DISTANCE FBOC	(in)		
DELAY TIME	(usec)		
HEAD #2: SIZE300 soft	(kV)		
DISTANCE FBOC	(in)		
DELAY TIME	(usec)		
HEAD #3: SIZE 300 soft	(KV)		
	(in)		
	(usec)		
VACUUM PRESSURE 4//2 (torr)			
TARGET DESCRIPTION MSFC LE 45			

COMMENTS NO PEN

INHIBITED S	HAPED CHARGE LAUNCHER DAT	A SHEET
TEST NUMBER	7698-/7 DATE	5-6-13
	/46	
EXPLOSIVE		
	ARGE NUMBER	_
CH	ARGE WEIGHT	_(grams)
DE'	TONATOR TYPE <u>RP-87</u>	_
INHIBITOR		
OV.	ERALL HEIGHT	_ (in)
INN		_ (in)
FXRs		
HEAD #1:	SIZE	_ (kV)
	DISTANCE FBOC	(in)
	DELAY TIME	_ (usec)
HEAD #2:	SIZE300 soft	
	DISTANCE FBOC	(in)
	DELAY TIME	_ (usec)
HEAD #3:	SIZE 300 soft	(kV)
	DISTANCE FBOC	(in)
	DELAY TIME	
VACUUM PRESS	SURE 4//2 (torr)	
TARGET DESCR	RIPTION MSFR. LOS	
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		E LAUNCHER D	
		DATE	
SCL TEST NO	/ \$ 7		noy \$ 75°
XPLOSIVE			
СНА	RGE NUMBER	<i>d</i> 5	age or produced and the
СНА	RGE WEIGHT	658.5	(grams)
DET	ONATOR TYPE	RP-87	A description of the second of
NHIBITOR		(
OVE	RALL HEIGHT	1.749	(in)
INNE	ER DIAMETER	0,468	(in)
XRs			
HEAD #1:	SIZE	150	(kV)
	DISTANCE FBO	S	(in)
	DELAY TIME	11	(usec)
HEAD #2:	SIZE	300 soft	(kV)
	DISTANCE FBOO	C	(in)
	DELAY TIME	10	(usec)
HEAD #3 :	SIZE	300 soft	(kV)
	DISTANCE FBO		(in)
	DELAY TIME		(usec)
ACUUM PRESSI	URE 4/2	(torr	
	PTION MSFC	1-65	

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NO PEN

- 15T Tesi Wint RP-80 Det.

		LAUNCHER DATA	
TEST NUMBER	7698-19	DATE <u>3</u>	-12-96
ISCL TEST NO	148	DATE <u>3</u> \$\lambda	my \$ 70°
EXPLOSIVE		// 	
CHA	RGE NUMBER	45	-
CHAR	RGE WEIGHT	659.0	(grams)
DETC	NATOR TYPE	RP-87	-
INHIBITOR			
OVER	RALL HEIGHT	1.750	_ (in)
INNE	R DIAMETER	0.468	_ (in)
FXRs			
	SIZE	150	_ (kV)
	DISTANCE FBOC		
	DELAY TIME	/1	_(usec)
HEAD #2:	SIZE		_(kV)
	DISTANCE FBOC	1/	_ (in)
	DELAY TIME	, ,	_(usec)
HEAD #3:	SIZE	300 soft	_(kV)
	DISTANCE FBOC		_ (in)
	DELAY TIME		_ (usec)
VACUUM PRESSU	RE 4/2	(torr)	
TARGET DESCRI	TON . ~		
	(WIT)-	A 0.050	FRONT FACE
COMMENTS			
COMMENTS	(,050)	118 ON /MLI] 1,86AP (,150 2219)

	APED CHARGE			
TEST NUMBER	7698-20	DAT	E3-/	2-76
ISCL TEST NO	149	modernica	5. ~,	y 70°
EXPLOSIVE		. ~		
CHAI	RGE NUMBER	1'/		
CHAI	RGE WEIGHTb	55.0		(grams)
DETC	NATOR TYPE	RP-87		
INHIBITOR		, 7 ~		
OVER	RALL HEIGHT	1, 750		· · ·
INNE	R DIAMETER	0.468		(in)
FXRs				
HEAD #1:	SIZE	150		(kV)
	DISTANCE FBOC	[]		(in)
	DELAY TIME	<u>U</u>		(usec)
HEAD #2:	SIZE	300 soft		(kV)
	DISTANCE FBOC_			
	DELAY TIME			
HEAD #3:	SIZE	300 soft		(kV)
	DISTANCE FBOC			
	DELAY TIME			
VACUUM PRESSU	RE 4/2		(torr)	
TARGET DESCRI	PTION MSFC	L45		

COMMENTS

		LAUNCHER DAT			
TEST NUMBER _	7698-21	DATE	?-13-76		
ISCL TEST NO	150	THE PARTY OF THE P			
EXPLOSIVE		1 7			
CHA	RGE NUMBER	/ >	-		
СНА	RGE WEIGHT	658.0	(grams)		
DET	ONATOR TYPE	RP-87	07 1440000		
INHIBITOR					
OVE	RALL HEIGHT	1.750	_ (in)		
INNE	ER DIAMETER	0,468	(in)		
FXRs					
HEAD #1:	SIZE	150	(kV)		
	DISTANCE FBOC	11	(in)	010	/
	DELAY TIME	11	(usec)		
HEAD #2:		300 soft		DID NOT DUMP	,
	DISTANCE FBOC		(in)	DOM	
	DELAY TIME	/1	(usec)		
HEAD #3:	SIZE	300 soft	(kV)		
	DISTANCE FBOC	11	(in)		
	DELAY TIME	222.7	(usec)		
VACUUM PRESSU	JRE <u>4//2</u>	(torr)			

COMMENTS

TARGET DESCRIPTION M SFC LEO. 8

INHIBITED SHAPE	ED CHARGE LAUNG	CHER DATA	A SHEET
TEST NUMBER 76	98-22	DATE	-13-76
ISCL TEST NO.	15]		
EXPLOSIVE			
CHARGE	NUMBER		
CHARGE	WEIGHT		(grams)
DETONA	TOR TYPE RP-87		
INHIBITOR			
OVERALI	HEIGHT		(in)
INNER DI	AMETER		(in)
FXRs			
HEAD #1: SIZ	Œ <u>150</u>		_ (kV)
DIS	STANCE FBOC	11	_ (in)
DE	LAY TIME	1 /	(usec)
HEAD #2: SIZ	TE300 soft		_ (kV)
DIS	STANCE FBOC		_ (in)
DE	LAY TIME		
HEAD #3: SIZ	7E 300 soft		(kV)
DIS	STANCE FBOC		_ (in)
	LAY TIME		(usec)
VACUUM PRESSURE			
TARGET DESCRIPTION	IN MSFC JE	M 45F	

COMMENTS

APPENDIX B

Projectile Geometry Measurements

Table 3. Projectile Geometry Measurements

TEST NO.	MASS (g)	VELOCITY (km/s)	DIAMETER (cm)	L/D	TOTAL ANGLE (°) (QUADRANT)
7698-1	1.35	11.68	0.871	1.4	26 (2)
7698-2	1.12	11.38	1.011	1.1	31 (2)
7698-3	0.99	11.64	0.623	2.3	40 (1)
7698-4	0.94	11.37	0.665	2.0	30 (2)
7698-5	1.03	11.77	0.696	1.9	54 (3)
7698-6	0.84	11.37	0.739	1.5	50 (2)
7698-7	0.99	11.37	0.820	1.2	33 (1)
7698-8	0.82	11.40			_
7698-9	0.97	11.42	0.831	1.1	61 (2)
7698-10	1.42	11.35	0.693	2.6	36 (1)
7698-11	1.28	11.47	0.602	3.4	7 (1)
7698-12	0.76	11.45	0.762	1	_
7698-13	1.04	11.51	0.742	1.3	27 (4)
7698-14			-		_
7698-15					_
7698-16	1.09	11.51	1.102	1	75 (4)
7698-17					_
7698-18	0.97	11.30	0.813	1.3	15 (4)
7698-19	1.11	11.32	0.716	2.0	15 (2)
7698-20	1.23	11.38	1.392	0.9	_
7698-21					_
7698-22	0.98	11.36	0.947	1.3	29 (1)

TEST N	UMBER	7698 - 1	/		
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	. 400	483	.092	11.66	1011
	.305	.70>	.074,.071	11.64	- 10,4
, 4	.408		.062 .031	11.68	+ 24.1
	. 258		, 069	11.70	

AVERAGE OUTER DIA	AMETER (in	0.343	ille (TTTTE Chilling Sound and Sound Sound and Sound Sou	
AVERAGE THICKNES				
INNER DIAMETER (in)				
LENGTH (in)			ne o organización de la contra per ej den	
L/D			caso labatico de francisco francisco que labacido	Michael Company (CT)
L/D	1,7	en de la comercia de	······································	
TOTAL ANGLE (deg)_	25.8	(Quadrant #)	2	
MASS (g)	1, 35			
AVERAGE VELOCITY		11.68		

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY: 2.7 g/cc

QUADRANT: Looking at Impact Surface

TEST N	UMBER 7	7698 - 2			
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	0.521	0.434	.076	11,38	7.0.12
	0.275	0,137	,065	11.36	- 30.0
4			, 073	11,39	+ 8.0
			, 068	11:38	7 0,0

AVERAGE OUTER DIA	METER (in	0.39	8	
AVERAGE THICKNESS			`.	
INNER DIAMETER (in)				
LENGTH (in)				
L/D				feend
TOTAL ANGLE (deg)		(Quadrant #)	2	
MASS (g)				
AVERAGE VELOCITY		11,38		*****

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY:

2.7 g/cc

QUADRANT: Looking at Impact Surface

2 1 3 4

TEST N	UMBER	7698 - 3	5		
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	,260		. 087, .086	11.65	
	. 239		, 074	11.63	+0
4	, 240	0,560	, 076	11,64	+ 39,7
	, 243		.066,.060	11.64	, , , , ,

AVERAGE OUTER DIAMETER (in) 0,246
AVERAGE THICKNESS (in) 0.074
INNER DIAMETER (in) 0.098
LENGTH (in) 0.560
L/D2.3
TOTAL ANGLE (deg) 39.7 (Quadrant #)
MASS (g) 0,99
AVERAGE VELOCITY (km/s) 11.64

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY: 2.7 g/cc
QUADRANT: Looking at Impact Surface

TEST N	TUMBER 7				
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	, 270		,079	11.40	19.5
	. 298	_	. 079	11.33	-19.5
4	. 206		, 093	11.38	+ 24.4
	, 232		, 061	_	'

AVERAGE OUTER DIAMETER (in)	0.262		
AVERAGE THICKNESS (in)	0.078		
INNER DIAMETER (in)			
LENGTH (in) 0.51			
L/D	and an activate an activate and activate activate and activate and activate activate and activate activate and activate activate activate and activate activ	According to the State of the S	and the state of t
_			
TOTAL ANGLE (deg) 29.9	(Quadrant #)		
MASS (g) 0.94			RODELLO SE
AVERAGE VELOCITY (km/s)	11.37		

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY: 2.7 g/cc
QUADRANT: Looking at Impact Surface

TEST NUMBER 7698-5 LENGTH VIEW OUTER VELOCITY YAW (view 3) WALL PITCH (view 4) DIAMETER (in) THICKNESS (km/s) (in) (deg) (in) 11,74 -24.5 11,80 4 11.76 276 -52.7 , 280 , , 267 068

AVERAGE OUTER D	DIAMETER (in)_	0.274			
AVERAGE THICKNESS (in) 0.07					
INNER DIAMETER (in)	32	enne provincia i marcini di dipologico proprio di socio di		
LENGTH (in)	0.515		700 - Constant Build all the authors provinces and		
L/D/.	88				
TOTAL ANGLE (deg)	543	(Quadrant #)	3		
MASS (g)	1.03		Martin Torres of Martin Indiana (Section Control Contr		
AVERAGE VELOCIT	'Y (km/s)	11,77			

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY:

2.7 g/cc

QUADRANT: Looking at Impact Surface

2 1 3 4

TEST N	UMBER	7698-6			
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	, 264		, 075	11,36	. ,
	, 271	, 389	,065	11.39	-48.6
4	. 330		,066	11,37	+ 20.9
	,298		_	_	, -

AVERAGE OUTER DIAMETER (in) 0.291
AVERAGE THICKNESS (in) 0.069
INNER DIAMETER (in) 0.153
LENGTH (in) 0.446
L/D
TOTAL ANGLE (deg) 50.1 (Quadrant #) 2
MASS (g) 0.84
AVERAGE VELOCITY (km/s)

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY: 2.7 g/cc
QUADRANT: Looking at Impact Surface

TEST N	IUMBER 7	1698-7			
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	. 328	.372	. 058	11.38	72 7
	. 322	,3/2	. 074	11.36	+ 32.2
4	. 320		.061	11.35	+ 10,5
	-		_	11,37	,

AVERAGE OUTER DIAMETER (in)_	0.323	
AVERAGE THICKNESS (in)	* *	Makingo Makingo and a
INNER DIAMETER (in)		
LENGTH (in) 0.372		
L/D /./5		
TOTAL ANGLE (deg) 33, 3	(Quadrant #)	
MASS (g) 0, 99		
AVERAGE VELOCITY (km/s)	/1, 37	

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY: 2.7 g/cc

QUADRANT: Looking at Impact Surface

TEST N	UMBER -				
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	_		0.071	11.46	
	_			11,36	-
4	_		0.064	11,44	_
	_	_	_	11.36	

AVERAGE THICKNES	SS (in)		
INNER DIAMETER (in)		
LENGTH (in)			
L/D			
TOTAL ANGLE (deg)_	(Quadrant #)		
MASS (g)	0,82		
AVERAGE VELOCITY	(km/s) 11,41	account of the state of the sta	our familians
	(+) is RIGHT	,	,
DENSITY	(+) is UP Y: 2.7 g/cc	2	1
QUADRA	ANT: Looking at Impact Surface	3	4

Broken Projectice

AVERAGE OUTER DIAMETER (in)___

7698 - 9 TEST NUMBER VIEW. LENGTH YAW (view 3) OUTER WALL **VELOCITY** DIAMETER (in) THICKNESS (km/s) PITCH (view 4) (deg) (in) (in) 3 , 284 11,42 .051 .061 ,365 -60.3 . 370 11,47 081,,063 4 11,37 + 16.9

AVERAGE OUTER DIAMETER (in)_	0.327
AVERAGE THICKNESS (in)	••
INNER DIAMETER (in) O.	199
LENGTH (in) 0. 365	
L/D	
TOTAL ANGLE (deg) 60.7	(Quadrant #) 2
MASS (g) 0.97	
AVERAGE VELOCITY (km/s)	11.42
(, , , , , , , , , , , , , , , , , , ,	

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY:

2.7 g/cc

QUADRANT: Looking at Impact Surface

2 1 3 4

TEST N	UMBER -	7678-10			
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	, 33 5		. 0 8 3	11.34) Q
		_	, 0 75	11,34	+ 35.9
4	. 273, 242	0.455	,060,.080	11, 36	.1 91
	, 256 , , 258	0,130	,064	_	+ 4.81

AVERAGE OUTER DIAMETER (in) 0, 2, 3
AVERAGE THICKNESS (in)	0.072
INNER DIAMETER (in)	0.129
LENGTH (in) 0,70	7
L/D	
TOTAL ANGLE (deg)	(Quadrant #)
MASS (g) /, 4 2	
AVERAGE VELOCITY (km/s)	11.35

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY:

2.7 g/cc

QUADRANT: Looking at Impact Surface

2 1 3 4

TEST NUMBER 7698-11 VIEW OUTER LENGTH WALL VELOCITY YAW (view 3) DIAMETER PITCH (view 4) THICKNESS (in) (km/s) (in) (deg) (in) 11,46 232 ,068 . 247 +5.51 ,057 11.49 ,197 4 + 4.65 11.48 .074 . 286 .771 11,46 . 231 . 069

AVERAGE OUTER DIAMETER (in)_	0,237
AVERAGE THICKNESS (in)	,
INNER DIAMETER (in)	
LENGTH (in) 0, 809	
L/D	
TOTAL ANGLE (deg) 7, 2	(Quadrant #)
MASS (g) /, 28	
AVERAGE VELOCITY (km/s)	11,47

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY:

2.7 g/cc

QUADRANT: Looking at Impact Surface

3

TEST N	UMBER	7698-1	2		
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	_			11.53	
		~	_	11,40	
4	_		_	11,48	
	_		_	11,39	

AVERAGE OUTER DIAMETER (in)		transamit o
AVERAGE THICKNESS (in)		
INNER DIAMETER (in)		nt/Stradificientessi@i
LENGTH (in)		
L/D		anguaga yan wanana
TOTAL ANGLE (deg) (Quadrant #)		
MASS (g) 0.76	en de la composite de la compo	and the same of th
AVERAGE VELOCITY (km/s) 11,45		
NOTES: YAW: (+) is RIGHT PITCH: (+) is UP DENSITY: 2.7 g/cc	2 3	1
QUADRANT: Looking at Impact Surface	3	4
Broken Prosectile		

TEST N	UMBER 7	698-13			
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	, 294		, 068	11.50	+ 12,8
	ر ۽ و ۲	_	,072	No.	+ / / / /
4	. 299	392	. 072	11.51	_ 24, 2
	, 308		,064		

AVERAGE OUTER DIAMETER (in	n) 0, 292
AVERAGE THICKNESS (in)	0.069
INNER DIAMETER (in)	0.154
LENGTH (in) 0.392	
L/D/, 3 4	
TOTAL ANGLE (deg) 26.7	(Quadrant #) 4
MASS (g) / 0 4	
AVERAGE VELOCITY (km/s)	11.50

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY: 2.7 g/cc

QUADRANT: Looking at Impact Surface

TEST N	UMBER	7692 - 11	6		
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3				11,51	
	_				
4	_			11.52	_
				11,5-1	

AVERAGE (OUTER DIAM	ETER (in)_			
INNER DIAM	METER (in)_				
LENGTH (in)				
L/D					
TOTAL AND	GLE (deg)	daystoon tare target the same and an arrange to the same and an arrange to the same and a same and	(Quadrant #)		on the second
MASS (g)	1,0	9			consideration all the
AVERAGE V	/ELOCITY (k	m/s)	11.51		one plante and the second
NOTES:	YAW: PITCH:	(+) is UP			
	DENSITY:		at Impact Surface	2	1
	201224			3	4

Broken Prosectice

TEST NUMBER 7698 - 18					
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	, 352	1 110 11	, 0 6 2	11.25	
	. 312	0.424	,078	11.22	+10.5
4	. 268		, 063	11,36	_ 10,5.
	. 348		, 057	11.35	-

AVERAGE OUTER DIAM	METER (in)	0.32	0	tandroopapaa
AVERAGE THICKNESS		4 2 -		
INNER DIAMETER (in)_		0.191		
LENGTH (in)				
L/D/, 3.3				
TOTAL ANGLE (deg)		(Ouadrant #)	4	
MASS (g)				
AVERAGE VELOCITY (k		11.30		bever/decilianshi

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY: 2.7 g/cc

QUADRANT: Looking at Impact Surface

TEST NUMBER 7698-19 VIEW OUTER LENGTH VELOCITY WALL YAW (view 3) DIAMETER (in) **THICKNESS** (km/s) PITCH (view 4) (in) (in) (deg) ,285 11.29 .064 -8,7 . 220 .063 4 , 324 . 077 11.35 0.560 + 12.5 .298 . 060

AVERAGE OUTER DIAMETER (in)0.28 2						
AVERAGE THICKNESS (in) 0.066							
INNER DIAMETER (in)	0.150						
LENGTH (in) 0, 5							
L/D							
TOTAL ANGLE (deg) /5.1	(Ouadrant #)						
MASS (g)/, //							
AVERAGE VELOCITY (km/s)	11,32						

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY:

2.7 g/cc

QUADRANT: Looking at Impact Surface

2 1 3 4

TEST N	UMBER -	7698-2	0		
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	0.548	0,473	.080	11.3/	
	-		_		4/3,8
4	_		_	11,46	-1819
			_	_	

AVERAGE OUTER DIAMETER (in)_	0,548	SSECONO conferencia de ser consensión con estreta que con estáblica potencia, se
AVERAGE THICKNESS (in)	1 052 1	
INNER DIAMETER (in)	0,388	
LENGTH (in) 0,47		
L/D 0.86		
TOTAL ANGLE (deg) 22,8	(Quadrant #)	/
MASS (g) 23		
AVERAGE VELOCITY (km/s)	11.38	

NOTES:

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY: 2.7 g/cc
QUADRANT: Looking at Impact Surface

TEST NUMBER 7698 - 22			2		
VIEW	OUTER DIAMETER (in)	LENGTH (in)	WALL THICKNESS (in)	VELOCITY (km/s)	YAW (view 3) PITCH (view 4) (deg)
3	, 410	.493	,07/	11.38	3.07.7
	, 336	,715			+28.6
4				11.35	+ 5.8

AVERAGE OUTER DIAMETER (in)_	0.373
AVERAGE THICKNESS (in)	
INNER DIAMETER (in)	1, 23,
LENGTH (in) 6,493)
L/D 1, 3 2	d ngila ar nga manananan nga ganga nga nga nga nga nga n
	1
TOTAL ANGLE (deg) 29,0	(Quadrant #)
MASS (g)	
AVERAGE VELOCITY (km/s)	11.36

N	0	T	E	C	
	1 /	18	8 °.	. 🤊	

YAW:

(+) is RIGHT

PITCH:

(+) is UP

DENSITY: 2.7 g/cc
QUADRANT: Looking at Impact Surface

3

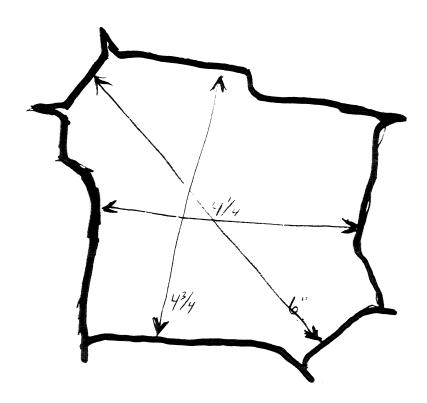
APPENDIX C

General Target Damage Measurements

1.25°

Volucing = 11.68 rm/s (high est?)

FACE



WALL

Winness

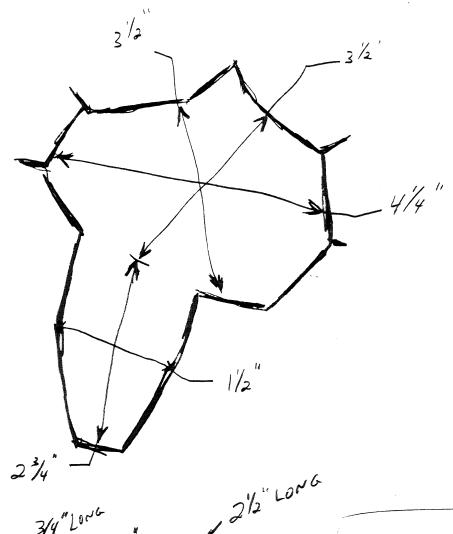
1 - Destaloyed

+, HURT BAD

= 3 - DAMAGED

7098-= LECO) verocity = 11,38 km/s (est)

FACE



WALL

1/4"

2/4"

2/4"

2/4"

2/4"

2/4"

1/4"

LONG

WITMOSS #1 4" \$ HOLE & A LOT OF TRASH

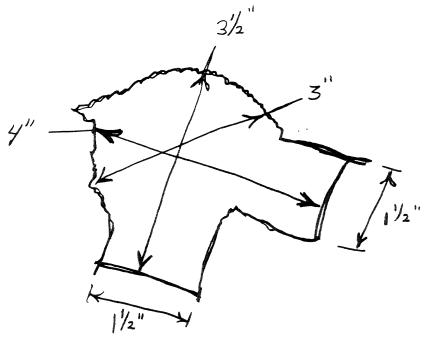
-21/4 LONG WIT # 2 | LARGO HAS \$ Trosh

WIT 43 3/4 \$ 100 Le

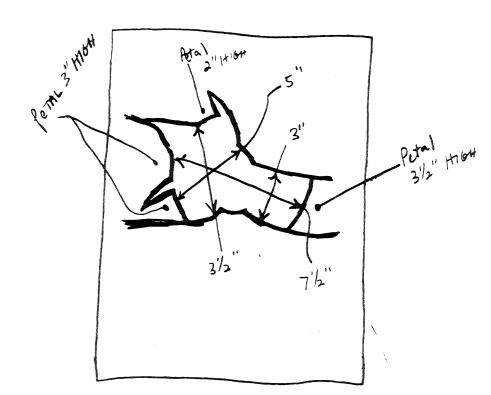
TARGET LEO Velocity = 11.64 km/s (ext.)

FACE

7698-3



WALL



WITNOSS OR 3

How MULTIPLE

SMAN TO MEDICA

(14"\$ TO 1"\$)

SIZE HOLES.

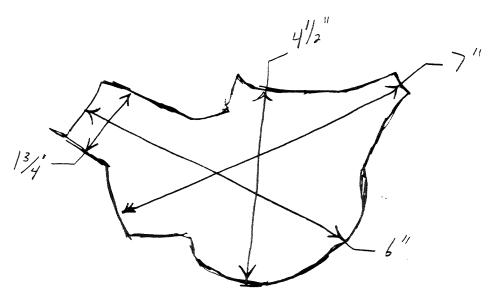
ALL 3 WERE

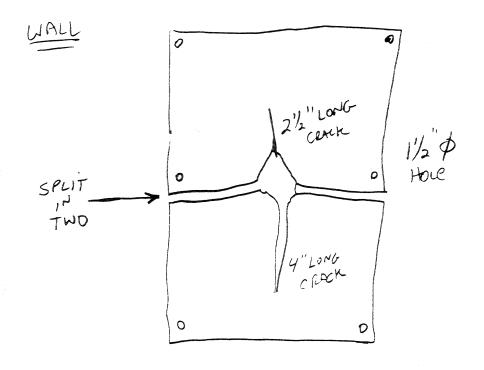
Deformed (Bert).

7698-4 L45

Vous i, = 11.38







WITTE

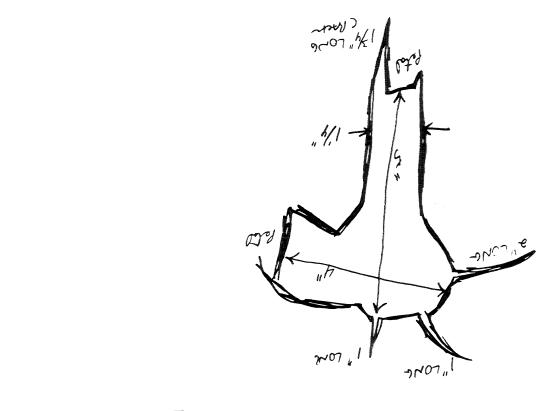
DAMAGE DECREASES FORM
MANY SMALL (1/2") NO =

ON PLATE # 1 70 ONLY

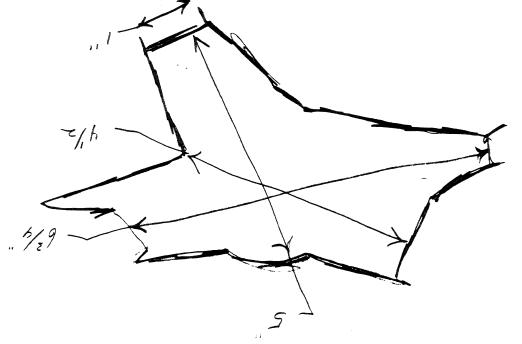
11/2 P

A FEW SMALL FORM
HOLE

(
//4" \$\phi\$) on free to



774M



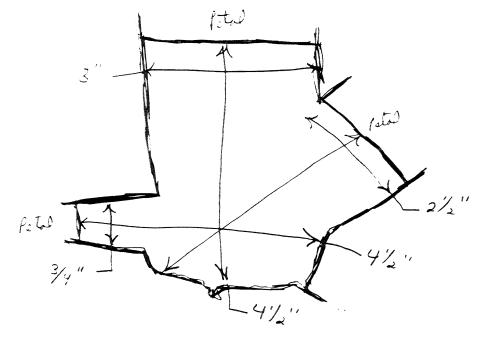
TENO

5-8696

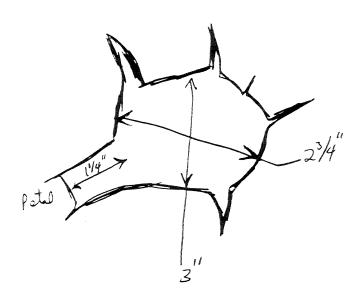
9 P. Mensiy

1673-6

D F, 16 F



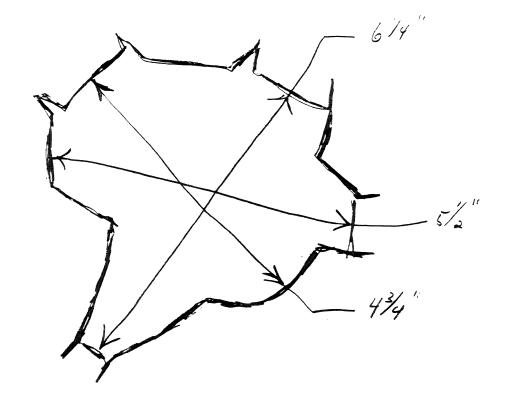
WALL



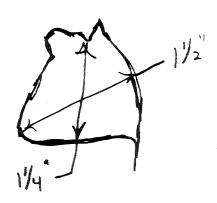
With house his not 1. December of In #=, Severel Smill from in #3 (May how his more and a Make Screen sould)

1078-7 LECY5

Fitz F

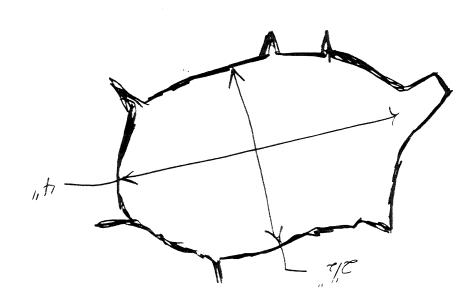


WALL

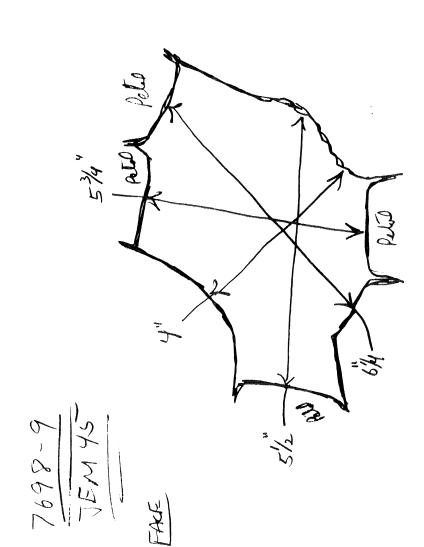


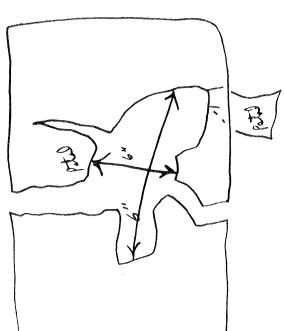
WITNESS #1 MOSTLY A COFTER SE OLUMINIM, 121, 12-1 Hours
#2 only very Familians (2/4/4)
#3 100 Yameroe

Frozic LAYRES THROUGH HOLE ~ 2-3" Pla



374



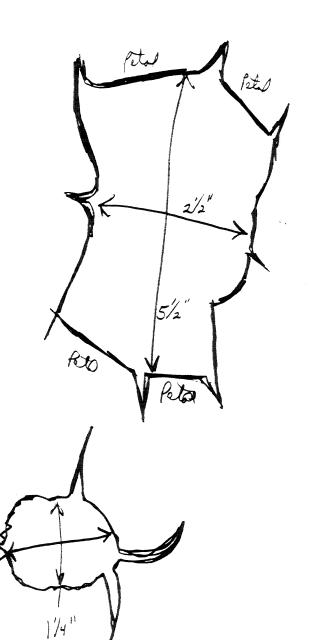


Single hole CITY #

(11/2, x3/9")

h t

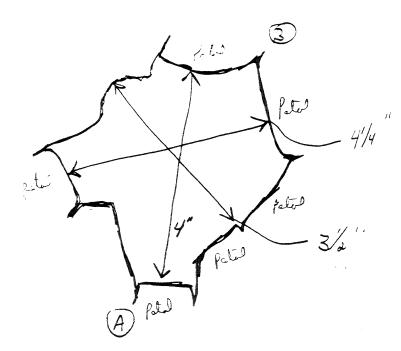
FACE



WAU

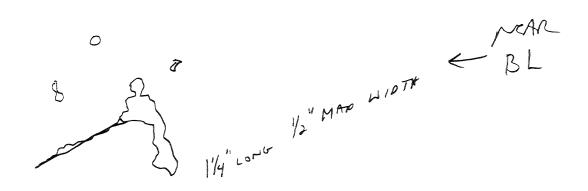
Witness #1 4.5 \$ Holo with MULTIME Holes all over # 2 \$ 73 Holo potter reduces to only small role by Rel =

78 28-11 LECOF



TIT = 43/4"

WALL

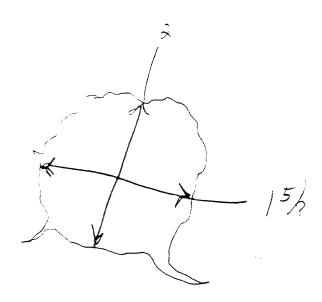


Witness #1 very little Deternation Several Smill hala

III 2 IMPACTS, NI PERETRATION

IFO. 7

FACE

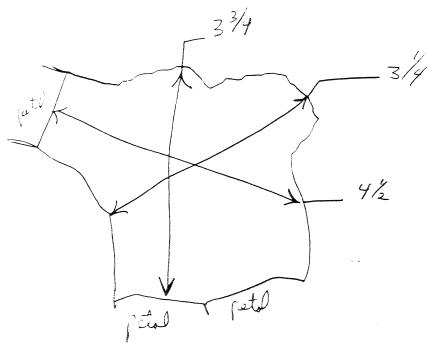


NALL NO HILC 1-11/4 BULDAE

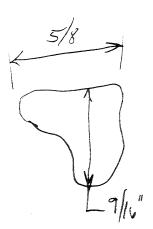
WITTAR NO DAMAGE

15.

The Frie



WALL



WITNESS - #1 HAS 2-3/4 / Hole = = 8 smller hole

d d ber 1-14-1/4 hole = = 1 cet 3/12 reig

= = = = = = = Note

7692-14

BAD CHARGE

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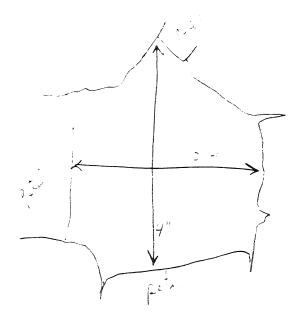
.

.

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LE 45

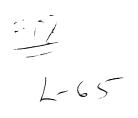
Down Fire



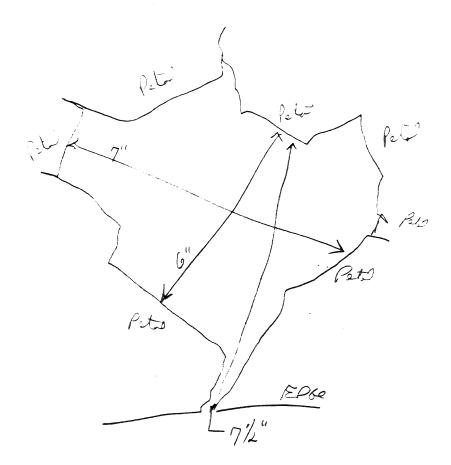
WALL - 110 FEN _3/4 BULGE

7678-17

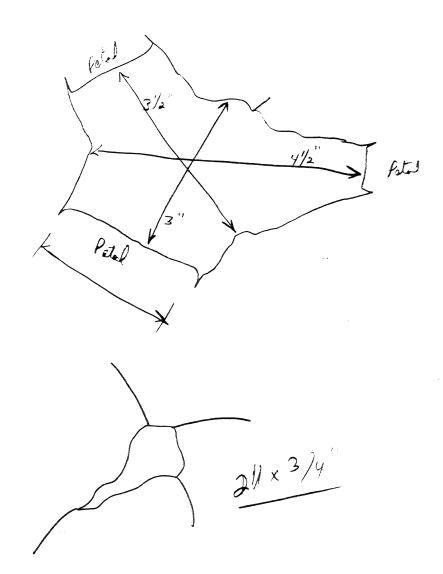
BAD CUANCE



FACE



FACE



Witte

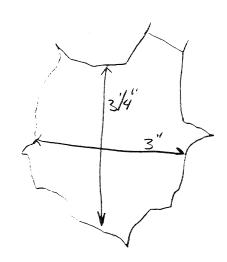
A FON Dents # only 1 Hole 154 #3

4/2 3 CE CE 10910 1/2 CO

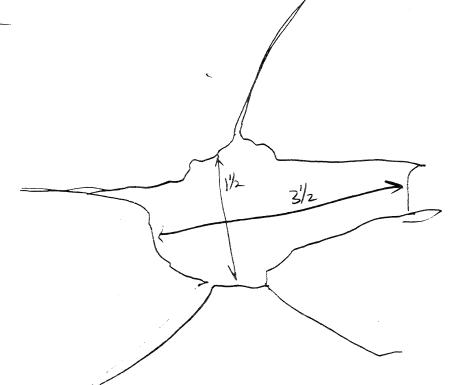
40 smod 700 1_3 SSONTIN)

7011-21 L=0.8

FACE

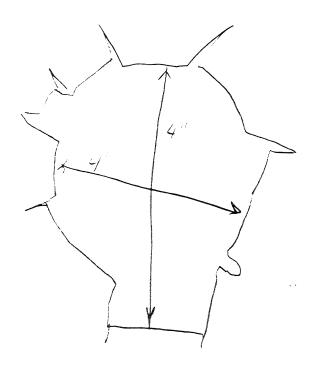


WALL

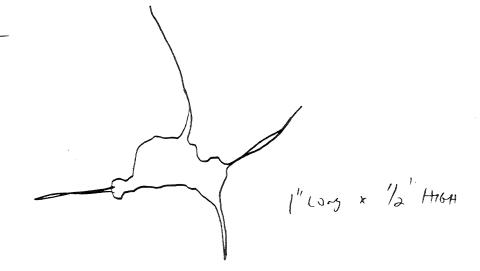


ADD 3 viction plater have Normerous Small holes

FASE



WALL



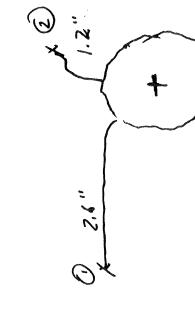
WITNESS 41 8-16 OR SMALLER HOTOS 1-3/9 \$ Acco

#13 1-1/4"\$ Hore

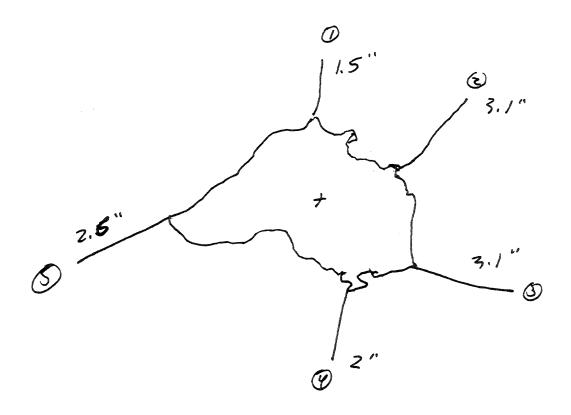
APPENDIX D MSFC Wall Hole Size Measurements

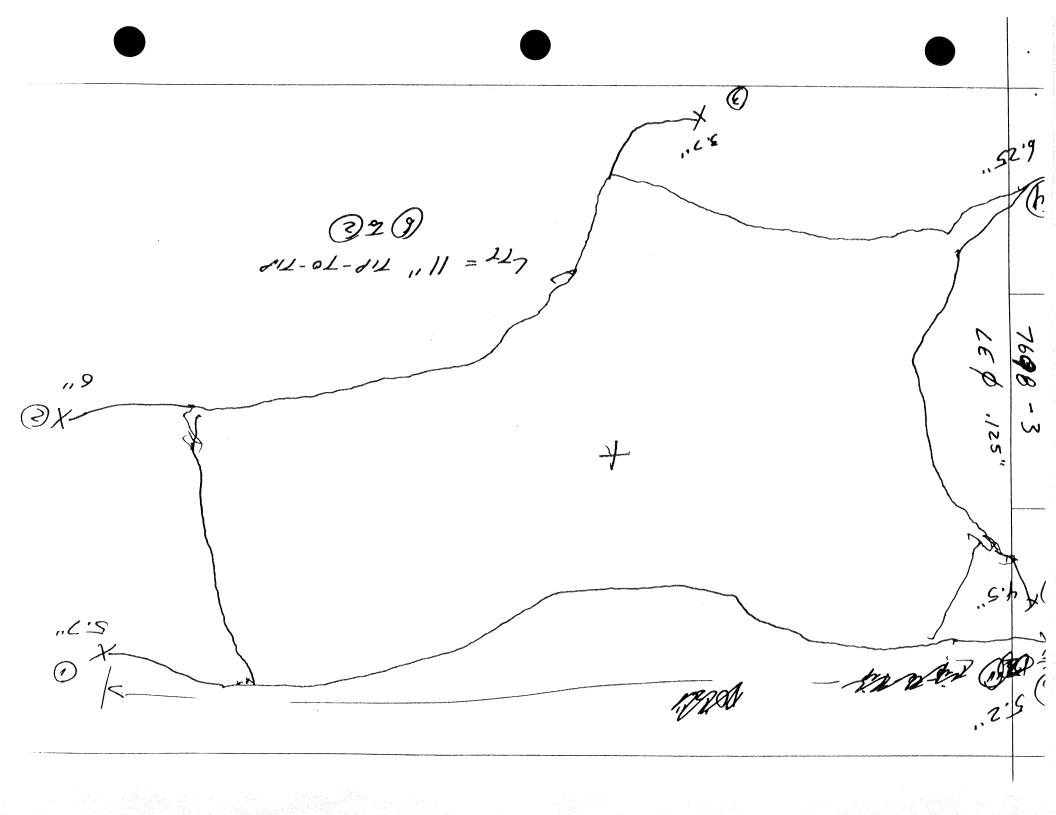
Table 4. MSFC Hole Size Measurements

TEST NO.	TIP-TO-TIP CRACK LENGTH (cm)
7698-1	6.86
7698-2	14.48
7698-3	27.94
7698-4	19.05
7698-5	21.08
7698-6	13.46
7698-7	9.14
7698-8	N/A
7698-9	33.02
7698-10	8.89
7698-11	7.94
7698-12	N/A
7698-13	1.75
7698-14	N/A
7698-15	N/A
7698-16	N/A
7698-17	N/A
7698-18	N/A
7698-19	12.07
7698-20	15.24
7698-21	17.78
7698-22	17.15

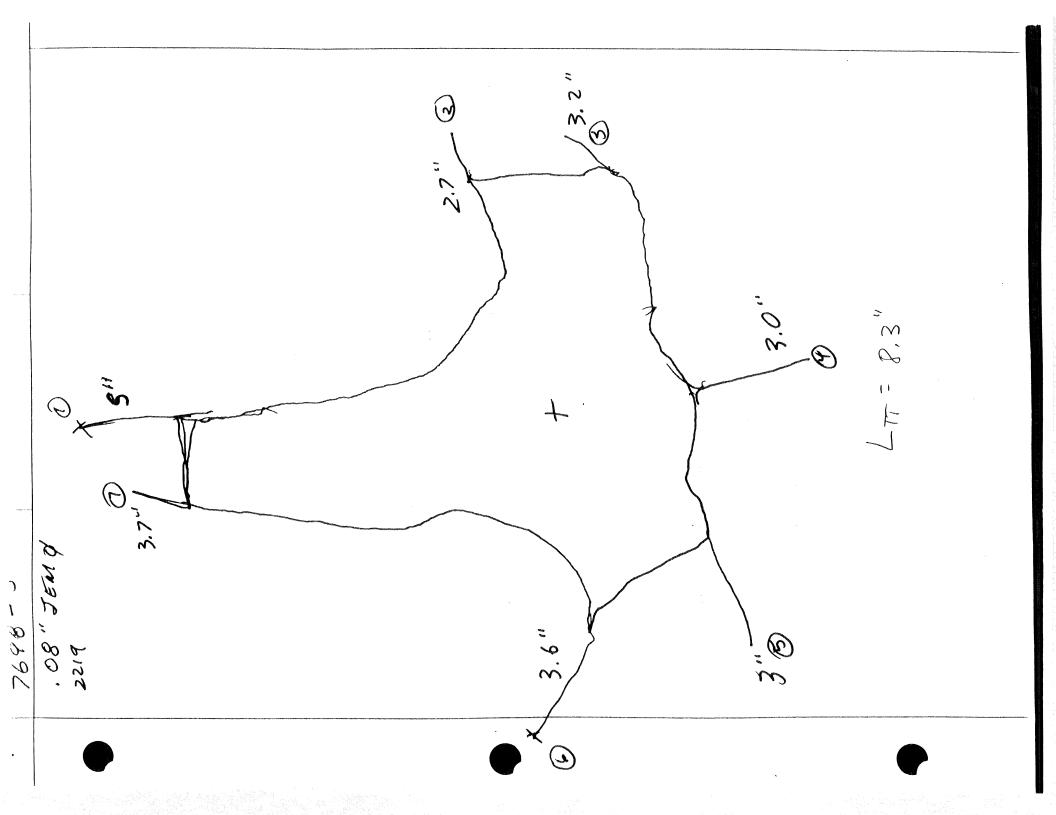


Lzy= 2,7"

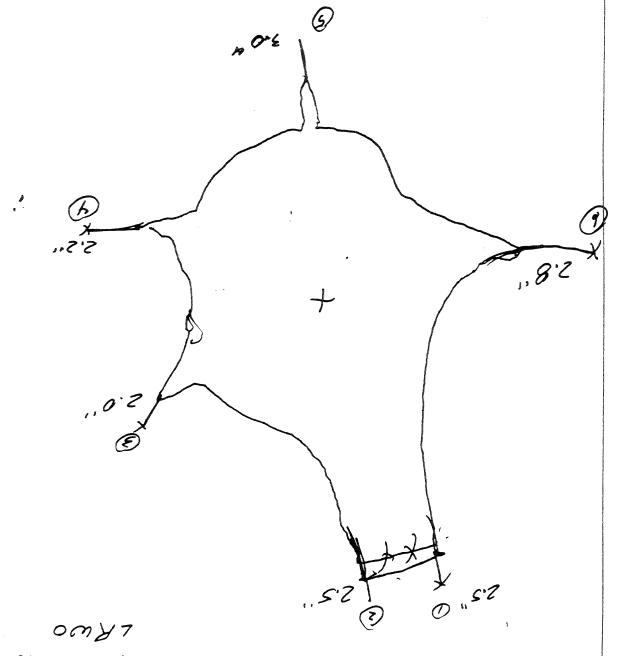




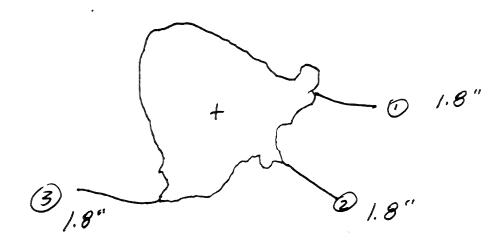
5/22 4-8692 .,52/.



15°5 = 147



9565 ,5211



LTT = 3.6"

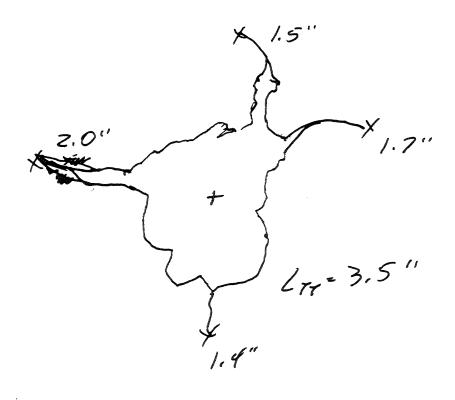
· 3070 J. 1.

2045 048

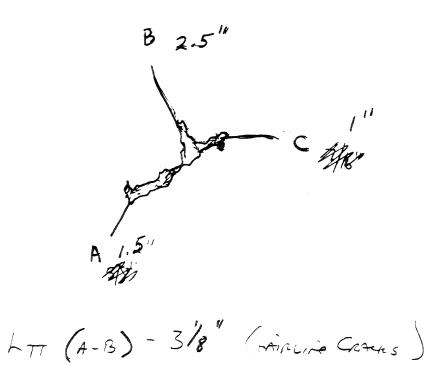
_Sh_37 b122 ,,521.

8-8/9/

.125" 5456 LRW 45



7698-11 LECOF



7698-12 BAD (SMALL, DIS-FORMED) PROJECTILE

1-1/2" BULDGE

7698-13 L45F

* \\ *

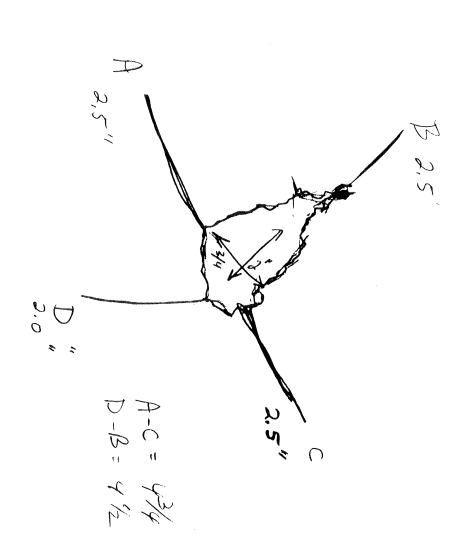
7098-14 - NO DET

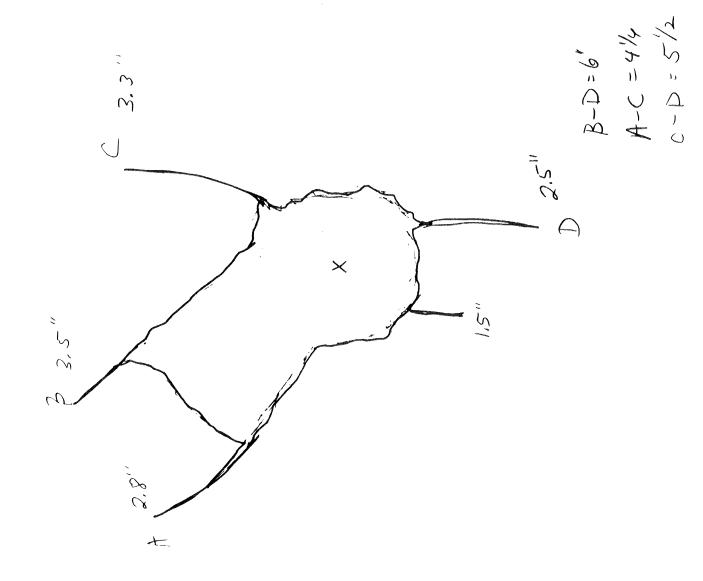
7698-15 - H- STRIFFIE

7698-16 - NO PEN

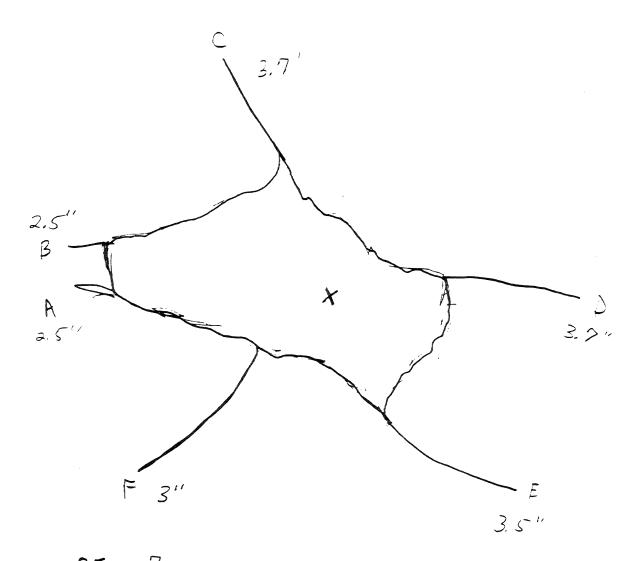
7098-17 - NO DET

7098-17 - NO DET

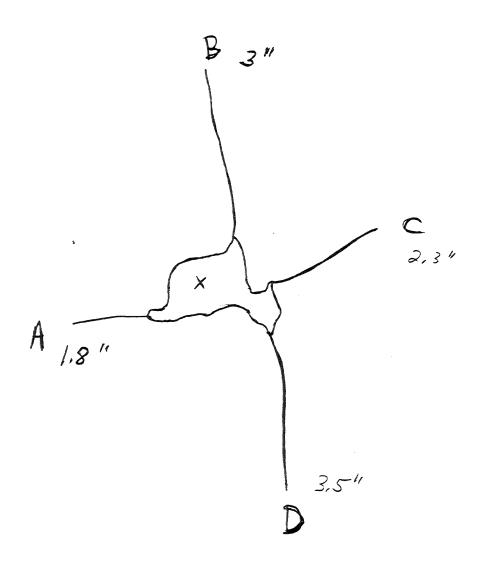




7698-21 LE0.8

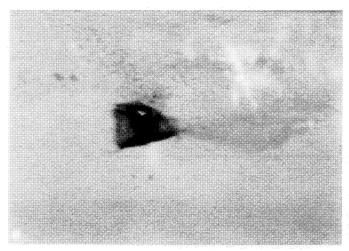


7698-22 JEM 45F



APPENDIX E

Radiographic Images



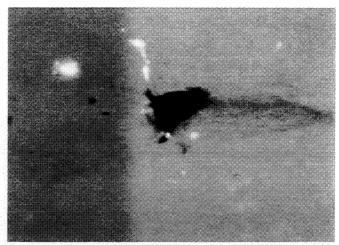
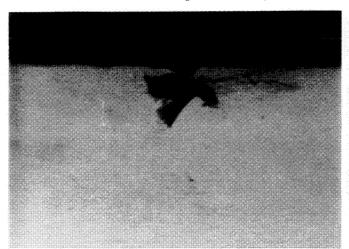


Figure 8. Projectile for Test 7698-1 (View 3 and 4).



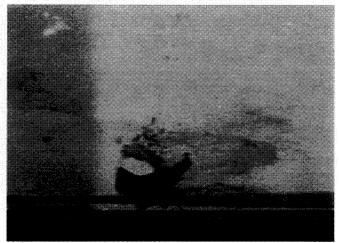


Figure 9. Projectile for Test 7698-2 (View 3 and 4).

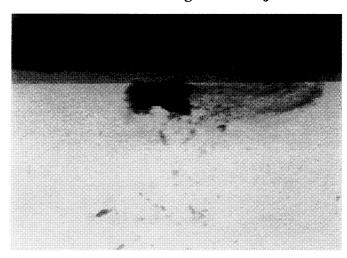




Figure 10. Projectile for Test 7698-3 (View 3 and 4).

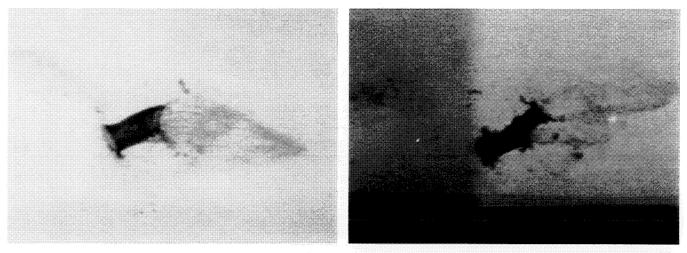


Figure 11. Projectile for Test 7698-4 (View 3 and 4).

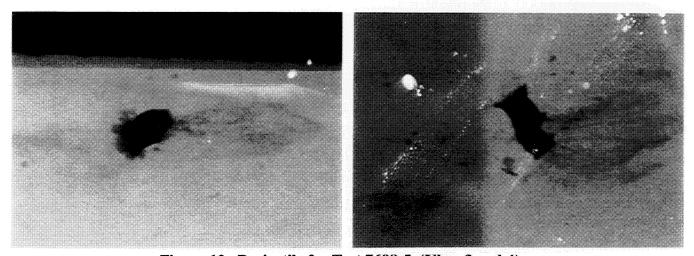


Figure 12. Projectile for Test 7698-5 (View 3 and 4).

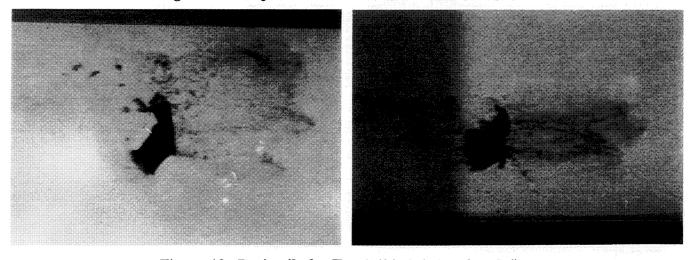


Figure 13. Projectile for Test 7698-6 (View 3 and 4).



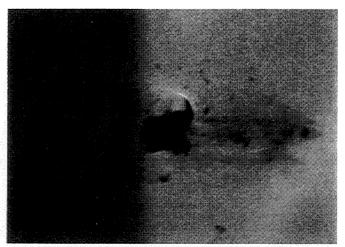


Figure 14. Projectile for Test 7698-7 (View 3 and 4).

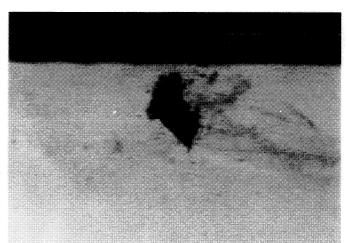
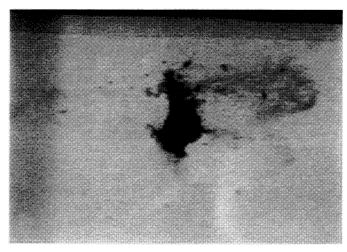




Figure 15. Projectile for Test 7698-8 (View 3 and 4).



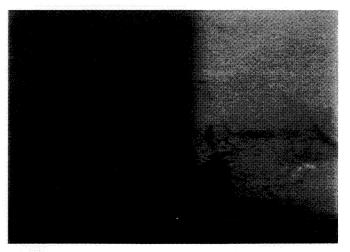


Figure 16. Projectile for Test 7698-9 (View 3 and 4).

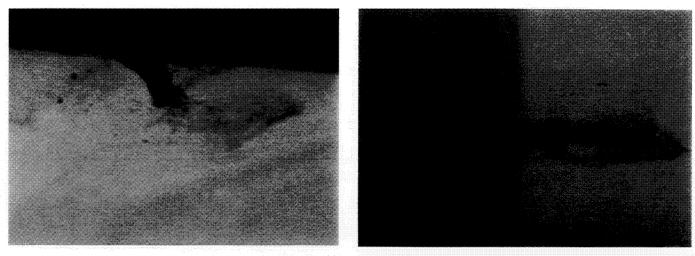


Figure 17. Projectile for Test 7698-10 (View 3 and 4).

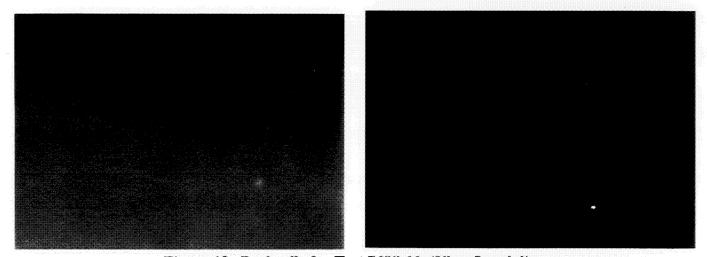


Figure 18. Projectile for Test 7698-11 (View 3 and 4).

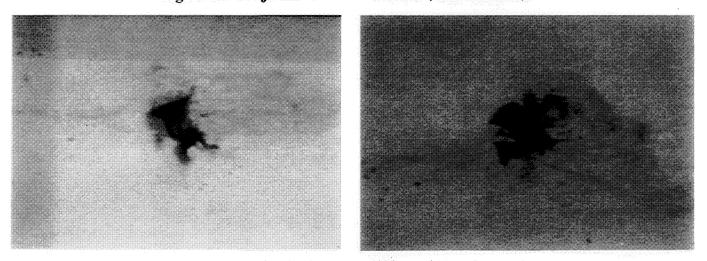
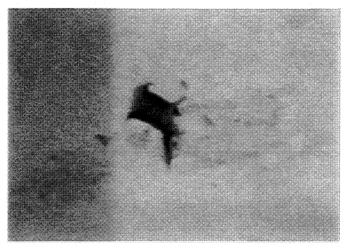


Figure 19. Projectile for Test 7698-12 (View 3 and 4).



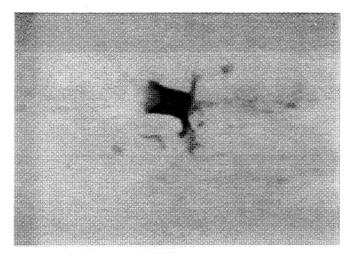
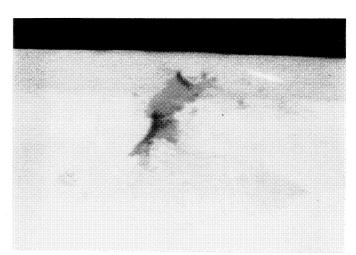


Figure 20. Projectile for Test 7698-13 (View 3 and 4).



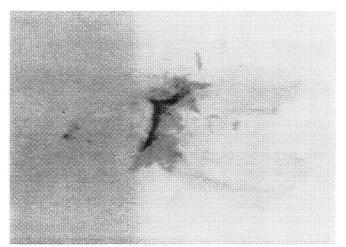
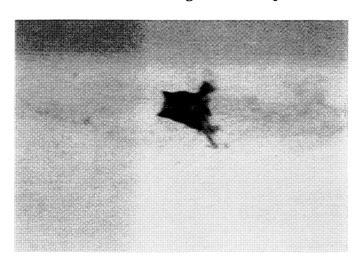


Figure 21. Projectile for Test 7698-16 (View 3 and 4).



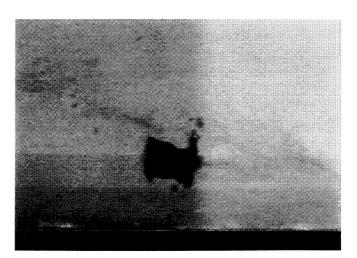
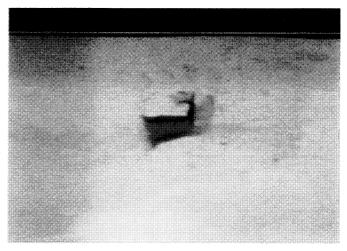


Figure 22. Projectile for Test 7698-18 (View 3 and 4).



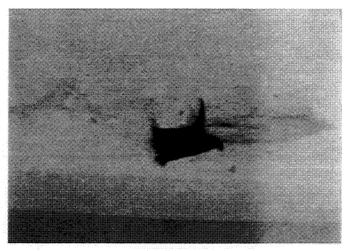


Figure 23. Projectile for Test 7698-19 (View 3 and 4).





Figure 24. Projectile for Test 7698-20 (View 3 and 4).

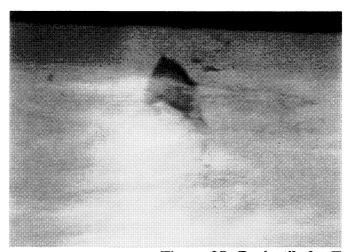




Figure 25. Projectile for Test 7698-22 (View 3 and 4).



Figure 26. Behind Wall Debris Pattern for Test 7698-1 (Horizontal View).

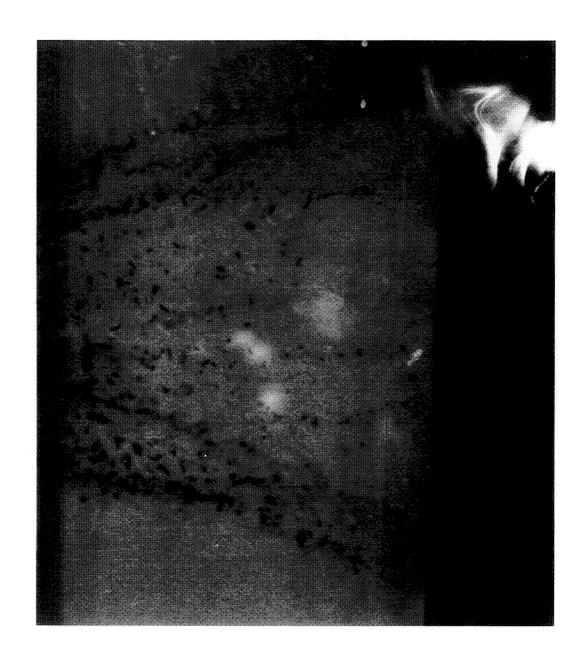


Figure 27. Behind Wall Debris Pattern for Test 7698-1 (Vertical View).

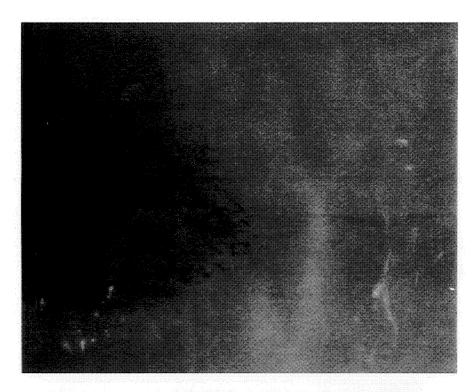


Figure 28. Behind Wall Debris Pattern for Test 7698-2 (Horizontal View).

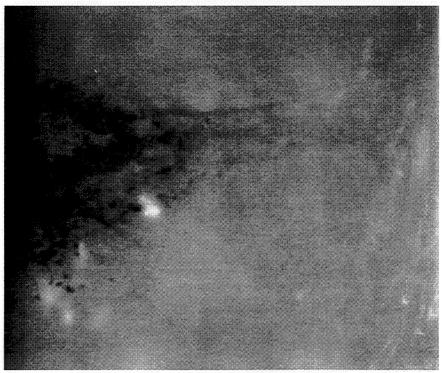


Figure 29. Behind Wall Debris Pattern for Test 7698-2 (Vertical View).

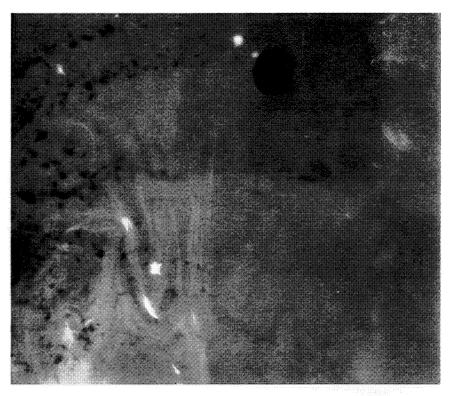


Figure 30. Behind Wall Debris Pattern for Test 7698-5 (Horizontal View).

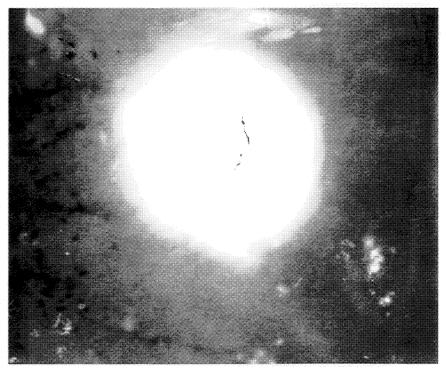


Figure 31. Behind Wall Debris Pattern for Test 7698-5 (Vertical View).



Figure 32. Behind Wall Debris Pattern for Test 7698-6 (Horizontal View).



Figure 33. Behind Wall Debris Pattern for Test 7698-6 (Vertical View).



Figure 34. Behind Wall Debris Pattern for Test 7698-11 (Horizontal View).

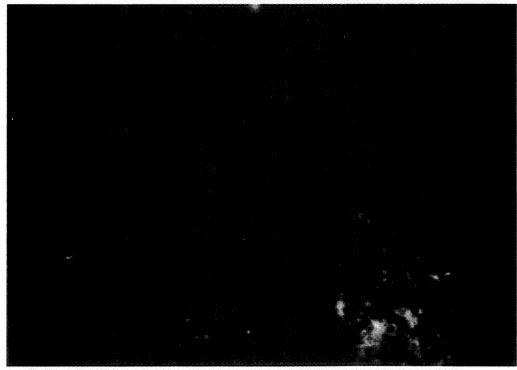


Figure 35. Behind Wall Debris Pattern for Test 7698-11 (Vertical View).



Figure 36. Behind Wall Debris Pattern for Test 7698-21 (Horizontal View).

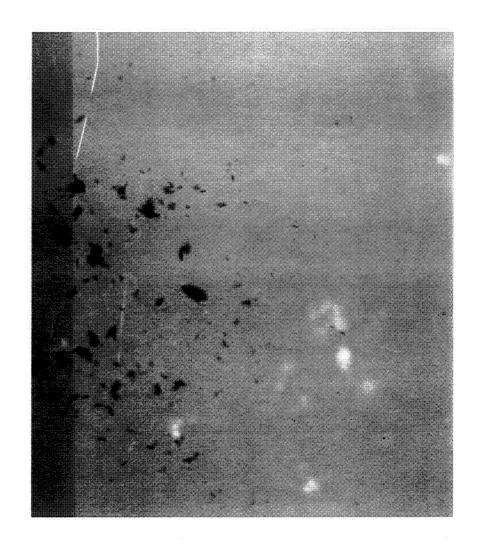


Figure 37. Behind Wall Debris Pattern for Test 7698-21 (Vertical View).

Appendix F
Test Summary for
Final Report 06-7139
dated October 1995

The work reported on in this test report (06-7698) is a continuation of work performed during SwRI Project Number 06-7139. Therefore, correlation of the data between these programs is essential. During the previous program, projectile mass calculations were made using a technique which has since been modified. Therefore, to insure that the data in this current work correlates properly with the data reported in the October 1995 report, the projectile masses from the 06-7139 report have been recalculated using the new calculation procedure. The following table reflects these changes.

Also, during the conduct of the 06-7139 experiments, are-calibration of the flash x-ray system occurred between Tests 7139-9 and 7139-10. It has since been determined, based on the extremely low velocity values measured during Tests 7139-10 through 7139-15, that this calibration was inaccurate. Therefore, the velocity values have been adjusted based on the average ISCL velocity. The procedure used to adjust the values is given below:

Average Velocity of "Slow" Projectiles: 10.69 km/s

Known Average ISCL Velocity: 11.28 km/s

Adjusted Value = ["slow" velocity - 10.69] + 11.28

Example: Test 7139-10:

Adjusted Velocity = [10.60 - 10.69] + 11.28 = 11.19 km/s

Table No. 2. Test Summary										
		Projectile Mass (g)	Projectile Velocity (km/s)	Target Description	Comments					
7139-1	5-2-95	1.24	N/M	U.S. Lab Whipple (0/)	Shield had a 1-1/2" diameter hole, Wall had a 1" diameter hole. All three witness plates were perforated.					
7139-2	5-2-95	1.30	11.14	Russian Whipple Shield (0/)	Shield had a 3" diameter hole, Wall had a 6" (post petal) diameter hole, and Plate #3 had a full waffle-square removed. All three witness plates were perforated.					
7139-3	5-3-95	1.30	11.16	ESA Cylinder (0/)	Shield had a 1-1/2" diameter hole, Wall had a 1-1/4" diameter hole, and Plate #3 had a 1-3/4" diameter hole. All three witness plates were perforated.					
7139-4	5-3-95	-	-	Enhanced Lab #1 (0/)	Projectile hit stripper plate. No data.					
7139-5	8-24-95	0.94	11.20	U.S. Lab Whipple (65/)	Shield had a 4" by 2" hole. MLI was destroyed. Material came off shield normal to its surface and missed wall. No witness damage.					
7139-6	8-25-95	1.06	11.18	ESA Cylinder (65/)	Shield had a 1-3/4" by 3" hole. Wall #1 had a narrow slit 2-1/4" long with several cracks forming. Wall #2 had not cracks or holes. No witness damage.					
7139-7	8-25-95	***	-	Jem Whipple (0/)	Projectile hit stripper plate. No data.					
7139-8	8-28-95	1.11	N/M	Jem Whipple (65/)	Shield had a jagged hole with major diameter of 3-1/2". MLI was destroyed. Wall had no cracks or holes. No witness damage.					

			Tab	le No. 2. Test Summary	
Test Number	Test Date	Projectile Mass (g)	Projectile Velocity (km/s)	Target Description	Comments
7139-9	8-28-95	0.88	N/M	Enhanced Lab #2 (0/)	Shield had a hole with major diameter of 2 1/2". Wall had a large, jagged hole with madiameter up to 5-1/2" and cracks up to 6" lo Large amount of pettaling of wall plate. A witness plates perforated.
7139-10	8-28-95	1.00	11.19*	Enhanced Lab #1 (0/)	Shield had a 3-1/2" diameter hole. The Nextlayer had a 3" diameter hole. The Kevlar la had a 5" diameter hole. The wall had a jagg hole with diameter up to 8-1/2" with sever cracks (one running to the plate edge). A witness plates perforated.
7139-11	8-29-95	1.20	11.26*	Enhanced Lab #1 (65/)	Shield had a 3-1/2" diameter hole. Nextel a Kevlar layers both had 3" to 4" diameter ho Wall had no holes but bulged 1". No witned damage.
7139-12	8-30-95	0.58 (est)	11.67*	Enhanced Russian (0/)	Projectile broke up. Shield had a 3" diame hole. Nextel and Kevlar layers had 3" diameter holes. Wall had a hole up to 7-1/ diameter.
7139-13	8-31-95	1.11	11.27*	U.S. Lab Whipple (45/)	Shield had a 4-3/4" by 2-1/2" diameter how Wall had only one small hole (1/2" diameter Witness 1 had a few small perforations, Witness 2 and 3 were not damaged.
7139-14	8-31-95	1.07 (est)	N/M	Russian Enhanced with Clamped Fabric (0/)	Shield had a 2-1/2" diameter hole. Nextel a Kevlar layers had 2" to 3" diameter holes Wall had a hole up to 7-1/2" diameter. Witness 1 had a 4" diameter hole. Witness and 3 had 2" to 3" diameter holes.

Shield had a 3" diameter hole. The Nextel and Kevlar layers had 4" diameter holes. The wall had a hole up to 5" diameter and several cracks (one extending to the plate edge).	Enhanced Lab #1 with Clamped Fabric (45/)	*10.11	\$6.0	\$6-\$-6	\$1-681 <i>L</i>					
Comments	Target Description	Projectile Velocity (km/s)	Projectile Mass (g)	Test Date	Test					
Table No. 2. Test Summary										